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STS-35 SCRUB 3 HYDROGEN LEAK ANALYSIS

By Dave Seymour

Propulsion Laboratory Science and Engineering Directorate

July 1991

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During the summer of orbiter disconnect hydrogen third scrub of STS-35, a leaf analysis of the data obtained prevalve was concluded to be observed leakage.	leak and multiple into k investigation team was plushing scrub 3 was p	ernal aft compartment vas organized. In supp erformed. Based on the	oort of this team, an his analysis, the engine 2
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TABLE OF CONTENTS

	Page
INTRODUCTION	1
SUMMARY	1
MODEL METHODOLOGY	1
SCRUB 3 MEASURED DATA	2
INITIAL MANIFOLD/ENGINE LEAK PREDUCTION RESULTS	3
SPECIAL TEST PREDICTION RESULTS	3
A. Recirculation Pump Backspin B. Prevalve Opening at 14,000 s (GMT 261:01:40) C. Initial Fast-Fill and Transition to Reduced Fast-Fill D. Prevalve Opening During Drain	4 4
PV5 SCRUB 3 PREDICTION RESULTS	5
PV5/PV6 SCRUB 2 PREDICTION RESULTS	6
PV5/PV6 TANKING TEST 1 PREDICTION RESULTS	7
PV5/PV6 SCRUB 1 PREDICTION RESULTS	7
CONCLUSIONS	7
APPENDIX	61

LIST OF ILLUSTRATIONS

Figure	Title	Page
1.	STS-35 S3 manifold pressure	17
2.	STS-35 S3 manifold temperature	18
3.	STS-35 S3 engine 1 inlet temperature	19
4.	STS-35 S3 engine 2 inlet temperature	20
5.	STS-35 S3 engine 3 inlet temperature	21
6.	STS-35 S3 engine 1 inlet pressure	22
7.	STS-35 S3 engine 2 inlet pressure	23
8.	STS-35 S3 engine 3 inlet pressure	24
9.	STS-35 S3 aft compartment high range H ₂ concentration	25
10.	STS-35 scrub 3	26
11.	STS-35 scrub 3	27
12.	STS-35 scrub 3	28
13.	STS-35 scrub 3	29
14.	STS-35 S3 aft compartment H ₂ concentration	30
15.	STS-35 S3 recirculation pump speed	31
16.	STS-35 S3 engine 1 inlet pressure	32
17.	STS-35 S3 engine 1 inlet temperature	33
18.	STS-35 S3 predicted aft compartment H ₂ concentration	34
19.	STS-35 S3 predicted aft compartment H ₂ concentration	35
20.	STS-35 S3 fast-fill/reduced-flow transition	36
21.	STS-35 S3 low range aft compartment H ₂ concentration	37
22.	STS-35 S3 manifold flow	38
23.	STS-35 S3 engine 2 recirculation pump vapor	39

LIST OF ILLUSTRATIONS (Continued)

Figure	Title	Page
24.	STS-35 S3 predicted PV5 leak	40
25.	STS-35 S3 predicted PV5 leak	41
26.	STS-35 S3 predicted PV5 leak	42
27.	STS-35 S3 predicted PV5 leak	43
28.	STS-35 S3 predicted PV5 leak	44
29.	STS-35 S3 predicted PV5 leak	45
30.	STS-35 S3 leak site subcool temperature	46
31.	STS-35 S3 80-percent PV5 + 20-percent 17-in line leak	47
32.	STS-35 S2 manifold pressure	48
33.	STS-35 S2 flow	49
34.	STS-35 S2 predicted versus actual aft H ₂ concentration	50
35.	STS-35 S2 leak site subcool versus manifold subcool	
36.	STS-35 T1 manifold pressure	
37.	STS-35 T1 flow	
38.	STS-35 T1 predicted versus actual aft H ₂ concentration	54
39.	STS-35 T1 leak site subcool versus manifold subcool	
40.	STS-35 S1 manifold pressure	56
41.	STS-35 S1 flow	. 57
42.	STS-35 S1 predicted versus actual aft H ₂ concentration	. 58
43.	STS-35 S1 leak site subcool versus manifold subcool	
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TECHNICAL MEMORANDUM

STS-35 SCRUB 3 HYDROGEN LEAK ANALYSIS

INTRODUCTION

Although considerable analysis had been performed on the previous STS-35 scrub and tanking test data, the wide range of operating conditions and valve configurations tested during scrub 3 was not previously available. With the formation of the STS-35 leak investigation team, an intensive analysis effort was undertaken to characterize the leak and determine its location. This report describes the analysis in roughly the chronological order in which it was performed.

The results of this analysis effort were provided to the leak investigation team as they became available. The results presented here sometimes differ slightly from those provided to the team during the investigation due to error corrections and refinements. However, the conclusions drawn from the results remain unchanged.

During the STS-35 scrub 3 leak isolation test, the behavior of the aft compartment H_2 concentration seemed to indicate a leak in the engine 2 prevalve (PV5). The decrease in H_2 concentration during drain associated with opening the engine 2 prevalve was particularly inditing. However, since the number of different leaks was unknown, all possible locations were considered in the analysis.

SUMMARY

Some of the more important conclusions drawn from this analysis were:

- 1. The leak did not exist at ambient temperature.
- 2. The engine 2 prevalve was the most likely leak location.
- 3. At least 80 percent of leakage came from the engine 2 prevalve.
- 4. The scrub 2 leak area was twice that of scrub 3 and consistent with the known engine 3 prevalve detent cover seal leakage.
- 5. Leak area changes cannot be inferred from concentration changes without employing an analysis similar to that used here.

MODEL METHODOLOGY

The basic approach taken was to calculate an aft compartment H_2 concentration based on the measured pressures, temperatures, and valve configurations and evaluate this prediction by direct comparison with the mass spectrometer data.

Prior to STS-35 scrub 3, H₂ flow rates had been calculated for an adiabatic real gas expansion over a range of initial pressures and temperatures. These data were converted to a table which provided relative flow rate as a function of the leak site pressure and degrees of subcooling.

Aft compartment H₂ concentration was calculated by assuming that the H₂ was uniformly mixed in a single volume and using the transient mass conservation equation:

$$\frac{d\,M_{\rm H_2}}{dt}=\dot{m}_{\rm H_{2_{\rm in}}}-\dot{m}_{\rm H_{2_{\rm out}}}$$

where,

 $M_{\rm H_2} = Y_c M_{\rm N_2}$

 $M_{\rm N_2}$ = mass of N₂ in aft compartment

 $\dot{m}_{\rm H_{2_{\rm in}}} = \rm H_2$ leak rate

 $\dot{m}_{\rm H_{2_{\rm out}}} = Y_c \, \dot{m}_{\rm N_2}$

 $\dot{m}_{\rm N_2}$ = aft compartment purge mass flow rate

 Y_c = aft compartment H_2 concentration.

While this approach preserves the basic time constant of the aft compartment purge flow (about 90 s), the calculated H_2 concentration begins to respond immediately to a change in leak flow rate. The discrete H_2 transport delay from the leak site to the vent door is smeared over the time constant. In addition, the predictions do not include time delays associated with the mass spectrometer H_2 concentration measurement.

SCRUB 3 MEASURED DATA

Rather than using the measured manifold pressure, a value calculated from measured ullage pressure and estimated liquid height was used. Both the measured and calculated manifold pressures are shown in figure 1.

The measured manifold temperature is compared with the saturation temperature corresponding to the calculated manifold pressure in figure 2. In this study, all saturation temperatures were determined from the pressure by the approximation:

$$TSAT(DEGR) = 9.96915 (psia)^{0.288438} + 15.0744$$
.

For leak rate predictions, temperature data were typically corrected for offsets by comparing the measurement to calculated saturation temperatures at appropriate times.

Engine inlet temperatures are compared with the calculated manifold saturation temperature in figures 3, 4, and 5. Engine inlet pressures are compared with the calculated manifold pressure in figures 6, 7, and 8.

The midrange 5,000 parts per million (p/m) aft compartment H_2 concentration measurements, GGDR2510T, was used to compare with all predictions. However, as shown in figure 9, there were occasional peaks above 5,000 p/m which were clipped by this measurement.

An event timeline was constructed which provides vehicle and facility valve actuation times and is shown in table 1. All data plots were made as seconds from the Space Transportation System (STS) data base Greenwich mean time (GMT) reference time of 90:260:21:47:00.0.

INITIAL MANIFOLD/ENGINE LEAK PREDICTION RESULTS

Initially, leak rate predictions were performed for the manifold based on the calculated manifold pressure and measured manifold temperature (V41T1428A). Predictions for each engine based on measured engine inlet pressure and measured inlet temperature were also performed. These results are compared with the 5,000 p/m range measured data in figures 10 to 13. These predictions did not include any heat leak effects on the leak site temperature.

The comparisons show the manifold prediction correlates much of the data except for the drain time period after 25,000 s. The engine 1 and engine 3 predictions show poor correlation. However, the engine 2 comparison shows good correlation during the drain period.

SPECIAL TEST PREDICTION RESULTS

After the initial manifold and engine predictions were performed, attention was focused on four specific time periods when unusual conditions and H_2 concentration behavior was observed.

A. Recirculation Pump Backspin

The first of these special conditions considered occurred between 9,000 and 10,000 s when, with engines 2 and 3 prevalves closed, the recirculation (recirc) valves were opened and engine 1 recirc began. The aft compartment H_2 p/m dropped as shown in figure 14. This sequence caused the engine 2 and engine 3 recirc pumps to backflow as shown by the recirc pump speed data in figure 15.

The engine 1 inlet pressure and temperature are shown in figures 16 and 17. Engines 2 and 3 inlet temperatures remained offscale high. The manifold temperature also increased during this time.

The recirc pump backspin was clearly the result of vapor flow from the warm engine inlet lines. Based on the recirc pump acceleration and estimated pump delta p, a vapor flow rate of 0.02 lb/s per pump was calculated.

Predicted leak rates were calculated for a leak site temperature corresponding to saturated liquid and saturated vapor when the recirc valve was open and the manifold temperature when closed. The comparisons in figure 18 show good agreement between the measured concentration and the saturated vapor prediction. However, a 120-s time delay was found which could not be totally explained.

The conclusion drawn from this analysis was that the vapor flowing out of recirc pumps 2 and 3 inlet produced vapor at the leak site. Since this occurred during reduced fast-fill and the flow patterns in the manifold were unknown, the leak could have been in the manifold, the prevalves, or the 17-in line.

B. Prevalve Opening at 14,000 s (GMT 261:01:40)

After engine 3 recirc was terminated and the recirc valves were closed, all three prevalves were opened and the p/m level dropped. Subsequent closing of the prevalves resulted in the p/m data returning to their previous level. During this period, the engine 1 bleed valve was closed and the inlet line dry, the engine 2 bleed valve was closed and the inlet line saturated, and engine 3 remained subcooled. The manifold temperature increased when engines 1 and 2 prevalves were opened.

Both engines 1 and 2 were probably producing vapor and a recirculating flow in their inlet lines. As in the previous special test analysis, a prediction was made using the manifold temperature when the engine 2 prevalve was closed and saturated liquid or vapor when the engine 2 prevalve was open. The comparison between the saturated vapor prediction and measured concentration shown in figure 19 indicated good agreement with about a 30-s time delay.

The conclusion drawn from this analysis was that the leak was probably not in the engine 3 prevalve, but further discrimination of the low pressure components was not possible.

C. Initial Fast-Fill and Transition to Reduced Fast-Fill

When the manifold flow rate was reduced from the 8,300 gallons per minute (gal/min) fast-fill rate to the 930 gal/min reduced fast-fill rate, the measured p/m dropped. About 10 min later, the concentration returned to the previous value.

The primary effect of manifold flow on the leak rate is the effect on leak site temperature. As a first approximation, the leak site temperature should be higher than the bulk fluid temperature by an amount proportional to the inverse of the flow rate. A prediction was made assuming a leak site temperature equal to the manifold temperature plus 0.6° during fast-fill and plus 6° during reduced fast-fill. The leak site temperature was arbitrarily returned to the manifold temperature at 3,400 s. The rise in leak rate at this time was thought to be the result of chill down and filling of the volume at the leak site and could not be reasonably modeled. Prior to 1,600 s, the predicted leak rate was set to zero. Figure 20 compares this prediction with the measured concentration.

Three conclusions were drawn from this analysis. The predicted 600 p/m from a saturated vapor leak was not observed in the data. Even a 200-p/m ambient temperature H_2 leak should have been detectable by the mass spectrometer. As shown in figure 21, the measured concentration was less than 30 p/m. This low measured concentration relative to the predicted value

confirmed that the effective leak area existed only when the leak site was chilled to operating temperatures and that an ambient temperature leak did not exist.

The second conclusion was that the manifold temperature and assumed heat leak effect was consistent with the measured concentration.

The third conclusion was that, with the prediction available, an ullage pressure effect on the leak rate could be seen in the measured concentration.

D. Prevalve Opening During Drain

During drain (starting at about 25,000 s), each engine's recirc pump was stopped and its prevalve opened. While there was no effect on the p/m data for engines 1 and 3, when engine 2's prevalve was opened the p/m level dropped. The engine 2 effect repeated three times.

As described in case b, each engine was probably producing vapor in its inlet line when its prevalve was open. As previously mentioned, the concentration data behavior was consistent with the engine 2 inlet temperature. Since the 5,700 gal/min drain flow produced a flow velocity of about 8 ft/s and there was no effect on the manifold temperature measurement and since the 17-in line is essentially horizontal, this vapor was very unlikely to rise above the manifold. Thus, these data excluded the 17-in line and the disconnect as possible leak sites. Further, it also excluded the engines 1 and 3 prevalves, since they produced no effect. These data, together with the recirc pump backspin data, indicated that the engine 2 prevalve, PV5, was the most likely leak location.

PV5 SCRUB 3 PREDICTION RESULTS

While the special test analysis had provided many important conclusions and given much insight into the leak behavior, it was felt that a unified model containing all the logic developed to date was required. This unified model would permit verification of all of the assumptions and logic through comparison with all of the concentration data.

The unified model consisted of a set of simple rules for determining the leak site temperature. The leak site temperature was taken to be the manifold temperature when the engine 2 prevalve was closed, otherwise it was taken to be the engine 2 inlet temperature. A heat leak effect was approximated by adding (930 gal/min/manifold flow) to this temperature. This may overestimate the heat leak effect but was judged acceptable. The manifold flow used is shown in figure 22.

To describe the effect of recirc pump vapor backflow, special logic was used to override the leak site temperature and instead used a saturated vapor leak rate. The logic used was that vapor backflow would occur when the recirc valve was open, when the recirc pump was off, when the prevalve was closed, and when the engine inlet temperature was superheated.

When this logic was initially implemented, the model predicted the concentration reduction previously analyzed. However, as shown in figure 23, the model also predicted backflow and concentration reduction at about 11,000 s which was previously unnoticed. In addition, when initially implemented, the logic was applied equally to all three engines. Engine 1 was predicted to

have vapor backflow between 21,100 and 12,600 s and between 12,900 and 14,100 s while no concentration reduction occurred in the data. Apparently, the local flow field transported the vapor from the engine 2 recirc inlet to the leak site but did not transport the vapor from recirc pump 1. This result was considered to be additional evidence of an engine 2 prevalve leak since each recirc pump is located directly in front of its prevalve. The final version of the model only applies the backflow logic to engine 2.

The model was run on a PC. A listing of the program is provided in the appendix. The model and all test data used in this analysis are available from the author on a PC floppy disk.

The model prediction is compared with the measured H_2 concentration in figure 24 for the full duration of the data and in figures 25 to 29 with an expanded time scale. Also shown in these figures are event bars which indicate when valves are open or the recirc pumps are on. Figure 30 shows the predicted leak site subcooling along with the manifold subcooling.

Two deviations of the prediction from the measured data are shown in figure 26. The first, at 11,400 s, occurs when closure of the recirc valves terminates vapor backflow throughout the engine 2 recirc pump and the model switches from a saturated vapor leak rate to a leak based on the subcooled manifold temperature. This deviation is thought to be caused by the time required to condense the accumulated vapor volume in the prevalve which is not modeled. A similar time delay was found when vapor backflow was terminated at 9,400 s.

The second deviation occurs at 12,700 s when the engine 2 prevalve is opened, and the model predicts a concentration decrease corresponding to the near saturated engine inlet temperature. The cause of this deviation is unknown. Similar prevalve openings between 14,000 and 15,000 s and during drain produced a concentration decrease as predicted.

To determine if all of the scrub 3 leak could be explained by one PV5 leak, several combinations of a PV5 leak and a manifold leak were predicted. As previously described and shown in figure 10, a pure manifold leak prediction does not include any effect of recirc pump vapor backflow or prevalve opening. Thus, combining the PV5 leak with a manifold leak tends to reduce the magnitude of the predicted concentration reduction for these effects. Figure 31 shows a prediction for an 80-percent PV5 leak combined with a 20-percent manifold leak. Further increases in the manifold leak fraction tends to raise the minimum concentration during the vapor backflow and PV5 open times to a value above the data. Therefore, the 20-percent manifold fraction was judged to be the highest reasonable fraction which could exist and most of the leak was in PV5.

PV5/PV6 SCRUB 2 PREDICTION RESULTS

To further verify the scrub 3 model, predictions were made for scrub 2. Figure 32 shows the calculated manifold pressure, and figure 33 shows the manifold flow rate. Initially, the scrub 3 model was applied to scrub 2 without any modification. Surprisingly, the predicted concentration was almost exactly half the measured data. Although the PV6 detent cover seal was a known additional leak source for scrub 2, the measured scrub 2 concentration was only slightly higher, and this result was totally unexpected.

Subsequent scrub 2 predictions modeled the PV6 leak as a separate second leak with the same leak area and same logic as developed for PV5. The final scrub 2 prediction is compared with the measured concentration in figure 34. The leak site subcooling and manifold subcooling

are shown in figure 35. The reasonably good agreement further verified the model but, more importantly, provided further evidence that most of the scrub 3 leak was due to the one PV5 leak.

These results clearly showed that the changes in concentration data between scrub 2 and scrub 3 could not be used to infer changes in leak area. The effects of the different valve configurations and different manifold flows could only be included by applying the model analysis. The scrub 3 leak area was actually half that of scrub 2.

PV5/PV6 TANKING TEST 1 PREDICTION RESULTS

This tanking test was primarily concerned with the orbiter/external tank disconnect leakage. High external H₂ concentrations resulting from the disconnect leakage prevented high fill rates. As shown in figure 37, the manifold flow rate was low for most of the test.

The model prediction H_2 concentration is compared with the measured data in figure 38. This prediction used a leak area 83 percent of that used for scrub 2. There is considerable uncertainty in the actual amount of leak area reduction since significant concentrations were obtained for only two brief periods.

PV5/PV6 SCRUB 1 PREDICTION RESULTS

The scrub 1 manifold flow conditions, shown in figure 41, were very similar to tanking test 1. Engine recirculation was not performed. The model prediction H₂ concentration is compared with measured data in figure 42 for a leak area of 69 percent of that used for scrub 2. Again, considerable uncertainty exists in the amount of this leak area reduction. However, since some additional H₂ leak flow into the aft compartment from the disconnect leak should have existed during scrub 1 and tanking test 1, there does appear to be a trend of increasing leak area with each loading.

CONCLUSIONS

The leak isolation procedures used during scrub 1 and tanking test 1 were designed to further understand the disconnect leak and did not produce useful data relative to an aft compartment leak. With sufficient amounts of the right test data, detailed analysis and modeling can determine the leak location and behavior. Although the results of the STS-35 analysis were not always sufficiently timely to significantly contribute to the leak investigation team's activities, future leak investigation analyses can build on the experience and analysis tools developed here.

Table 1. STS-35 scrub 3 LH₂ leak isolation timeline.

REMARKS - PROCEDURE:					REDUCED FLOW RATE TO	930 GPM	ISOLATE LH2 MANIFOLD				ISOLATE HI PT BLEED	LINE		ISOLATE 4-INCH LINE	AND DISCONNECT						
DESCRIPTION	A3309 CHLDWN VLV OPEN #1 IND CHILLDOWN COMPLETE	SLOW FILL TO 2% COMPLETE	MPS LH2 TOPPING VLV (PV13) OP IND	MPS LH2 HI PT BL VLV (PV22) OP IND	A3301 XFR LINE VLV CLOSED #1 IND		MPS E1 LH2 PREVLV (PV4) CL IND	MPS E2 LH2 PREVLV (PV5) CL IND	MPS E3 LH2 PREVLV (PV6) CL IND	MPS LH2 TOPPING VLV (PV13) CL IND	MPS LH2 HI PT BL VLV (PV22) CL IND		MPS LH2 HI PT BL VLV (PV22) OP IND	MPS LH2 TOPPING VLV (PV13) OP IND		MPS LH2 INBD F/D VLV (PV12) CL IND	MPS LH2 HI PT BL VLV (PV22) CL IND	MPS E1 LH2 PREVLY (PV4) OP IND A		E3 LH2 PREVLY (PV6) OP TND	E1 LH2 PREVLV (PV4) CL IND
EVENT	OPEN	NO	OPEN	OPEN	CLOSED		CIOSED	CLOSED	CLOSED	CLOSED	CLOSED		OPEN	OPEN		CLOSED	CLOSED	OPEN	OPEN	OPEN	CLOSED
TIME (SEC)	311.007	1200.008	2009.015	2122.015	2590.015		3939.023	3946.023	3950.023	3955.023	5007.027		5946.035	5960.035		5974.035	6256.035	7578.043	7582.043	7588.043	7730.043
GMT_TIME	260:21:52:11: 6 260:21:56:53: 0	260:22:20: 2: 0	260:22:20:29: 14	260:22:22:22: 14	260:22:30:10: 14		260:22:52:39: 22	260:22:52:46: 22	260:22:52:50: 22	260:22:52:55: 22	260:23:10:27: 27		260:23:26: 6: 35	260:23:26:20: 35		260:23:26:34: 35	260:23:31:16: 35	260:23:53:18: 42	260:23:53:22: 42	260:23:53:28: 42	260:23:55:50: 42

Table 1. STS-35 scrub 3 LH₂ leak isolation timeline (continued)

			ISOLATE ENGINE 1								BACKSPIN	BACKSPIN													
MPS E2 LH2 PREVLV (PV5) CL IND	MPS E3 LH2 PREVLV (PV6) CL IND	MPS LH2 HI PT BL VLV (PV22) OP IND	MPS LH2 INBD F/D VLV (PV12) OP IND	MPS LH2 TOPPING VLV (PV13) CL IND	MPS E1 LH2 PREVLV (PV4) OP IND A	MPS E1 LH2 RECIRC VLV(PV14) OP IND	MPS E3 LH2 RECIRC VLV(PV16) OP IND	MPS E2 LH2 RECIRC VLV(PV15) OP IND	MPS E1 LH2 RECIRC PUMP (PP1) SPEED	MPS E1 LH2 PREVLV (PV4) CL IND	MPS E2 LH2 RECIRC PUMP (PP2) SPEED	MPS E3 LH2 RECIRC PUMP (PP3) SPEED	MPS E1 LH2 RECIRC PUMP (PP1) SPEED	MPS E1 LH2 RECIRC VLV (PV14) CL IND	MPS E2 LH2 RECIRC VLV (PV15) CL IND	MPS E3 LH2 RECIRC VLV (PV16) CL IND	MPS E2 LH2 RECIRC PUMP (PP2) SPEED	MPS E3 LH2 RECIRC PUMP (PP3) SPEED	MPS E1 LH2 PREVLV (PV4) OP IND A	E1 LH2 BLEED VALVE	E1 LH2 BLEED VALVE	E2 LH2 BLEED VALVE	E3 LH2 BLEED VALVE	MPS E1 LH2 RECIRC VLV(PV14) OP IND	MPS E3 LH2 RECIRC VLV(PV16) OP IND
CLOSED	CLOSED	OPEN	OPEN	CLOSED	OPEN	OPEN	OPEN	OPEN	NO	CLOSED	NO	NO	OFF	CLOSED	CLOSED	CLOSED	OFF	OFF	OPEN	CLOSED	OPEN	CLOSED	CLOSED	OPEN	OPEN
7734.043	7738.043	8077.043	8543.047	8546.047	9245.051	9258.051	9258.051	9259.051	9267.051	9276.051	9356.051	9364.051	9373.051	9386.055	9386.055	9387.055	9387.055	9387.055	10211.059	10221.000	10242.000	10696.000	10706.000	10712.059	10712.059
260:23:55:54: 42	260:23:55:58: 42	261: 0: 1:37: 42	261: 0: 9:23: 46	261: 0: 9:26: 46	261: 0:21: 5: 50	261: 0:21:18: 50	261: 0:21:18: 50	261: 0:21:19: 50	261: 0:21:27: 50	261: 0:21:36: 50	261: 0:22:56: 50	261: 0:23: 4: 50	261: 0:23:13: 50	261: 0:23:26: 54	261: 0:23:26: 54	261: 0:23:27: 54	261: 0:23:27: 54	261: 0:23:27: 54	261: 0:37:11: 58	261: 0:37:21: 0	261: 0:37:42: 0	261: 0:45:16: 0	261: 0:45:26: 0	261: 0:45:32: 58	261: 0:45:32: 58

Table 1. STS-35 scrub 3 LH₂ leak isolation timeline (continued)

										ISOLATE ENGINE 2															
MPS E2 LH2 RECIRC VLV (PV15) OP IND	MPS E1 LH2 RECIRC PUMP (PP1) SPEED	MPS E1 LH2 PREVLV (PV4) CL IND	MPS E1 LH2 RECIRC PUMP (PP1) SPEED	MPS E2 LH2 RECIRC VLV (PV15) CL IND	MPS E3 LH2 RECIRC VLV (PV16) CL IND	E1 LH2 BLEED VALVE	MPS E1 LH2 RECIRC VLV (PV14) CL IND	E1 LH2 BLEED VALVE	MPS E1 LH2 PREVLV (PV4) OP IND A	MPS E2 LH2 PREVLV (PV5) OP IND A		MPS E1 LH2 PREVLV (PV4) CL IND	MPS E1 LH2 RECIRC VLV (PV14) OP IND	MPS E2 LH2 RECIRC VLV (PV15) OP IND			MPS E2 LH2 RECIRC PUMP ,(PP2) SPEED	MPS E2 LH2 PREVLV (PV5) CL IND	MPS E2 LH2 RECIRC PUMP (PP2) SPEED	MPS E1 LH2 RECIRC VLV (PV14) CL IND	MPS E2 LH2 RECIRC VLV(PV15) CL IND	MPS E3 LH2 RECIRC VLV (PV16) CL IND)	E2 LH2 BLEED VALVE	MPS E2 LH2 PREVLV (PV5) OP IND A
OPEN	NO	CLOSED	OFF	CLOSED	CLOSED	CLOSED	CLOSED	OPEN	OPEN	OPEN	OPEN	CLOSED	OPEN	OPEN	OPEN	CLOSED	NO	CLOSED	OFF	CLOSED	CLOSED	CLOSED	CLOSED	OPEN	OPEN
10713.059	10805.059	10813.059	11392.062	11398.062	11399.062	11401.000	11404.062	11425.000	11443.062	11646.062	11665.000	11710.062	12108.066	12108.066	12108.066	12121.000	12126.066	12133.066	12658.070	12664.070	12664.070	12664.070	12671.000	12680.000	12681.070
261: 0:45:33: 58	261: 0:47: 5: 58	261: 0:47:13: 58	261: 0:56:52: 62	261: 0:56:58: 62	261: 0:56:59: 62	261: 0:57:01: 0	261: 0:57: 4: 62	261: 0:57:25: 0	261: 0:57:43: 62	261: 1: 1: 6: 62	261: 1: 1:25: 0	261: 1: 2:10: 62	261: 1: 8:48: 66	261: 1: 8:48: 66	261: 1: 8:48: 66	261: 1: 9: 1: 0	261: 1: 9: 6: 66	9	61: 1:17:58:		261: 1:18: 4: 70	261: 1:18: 4: 70	261: 1:18:11: 0	261: 1:18:20: 0	261: 1:18:21: 70

Table 1. STS-35 scrub 3 LH₂ leak isolation timeline (continued)

ISOLATE ENGINE 3				ISOLATE RECIRC LINES
MPS E3 LH2 PREVLV (PV6) OP IND A E3 LH2 BLEED VALVE E2 LH2 BLEED VALVE	E1 LH2 PREVLV (PV4) OP IND E1 LH2 PREVLV (PV4) CL IND E1 LH2 RECIRC VLV(PV14) OP	MPS E2 LH2 RECIRC VLV(PV15) OP IND MPS E3 LH2 RECIRC VLV(PV16) OP IND MPS E1 LH2 RECIRC PUMP (PP1) SPEED MPS E1 LH2 RECIRC PUMP (PP1) SPEED MPS E2 LH2 PREVLV (PV5) CL IND	IND CL CL CL CL	E2 E1 E1 E1 E1 E2 E1 E2 E1 E1 E1 E1 E1 E1 E1
OPEN OPEN CLOSED	OPEN CLOSED OPEN	OPEN OPEN ON OFF	CLOSED OFF CLOSED CLOSED CLOSED CLOSED	OPEN OPEN OPEN OPEN CLOSED CLOSED
12731.070 12767.000 12824.000	850. 862. 886.	12886.070 12886.070 12886.070 12905.070 12961.070	13499.074 13497.074 14059.078 14068.078 14069.078	031. 1139. 143. 189. 196. 232.
261: 1:19:11: 70 261: 1:19:47: 0 261: 1:20:44: 0	1: 1:21:10: 1: 1:21:22: 1: 1:21:46:	261: 1:21:46: 70 261: 1:21:46: 70 261: 1:21:46: 70 261: 1:22: 5: 70 261: 1:23: 1: 70	261: 1:31:49: 74 261: 1:31:57: 74 261: 1:41:19: 78 261: 1:41:28: 78 261: 1:41:28: 78 261: 1:41:29: 78 261: 1:41:40: 0	1: 1:42:10: 1: 1:42:39: 1: 1:42:43: 1: 1:43:29: 1: 1:43:36: 1: 1:44:12: 1: 1:44:17:

Table 1. STS-35 scrub 3 LH₂ leak isolation timeline (continued)

MPS E3 LH2 PREVLV (PV6) CL IND	MPS E1 LH2 PREVLV (PV4) OP IND A	MPS E2 LH2 PREVLV (PV5) OP IND A	MPS E3 LH2 PREVLV (PV6) OP IND A	MPS E1 LH2 RECIRC VLV(PV14) OP IND	MPS E3 LH2 RECIRC VLV (PV16) OP IND	MPS E2 LH2 RECIRC VLV (PV15) OP IND	MPS E1 LH2 RECIRC PUMP (PP1) SPEED	MPS E2 LH2 RECIRC PUMP (PP2) SPEED	MPS E3 LH2 RECIRC PUMP (PP3) SPEED	MPS E1 LH2 PREVLV (PV4) CL IND	MPS E2 LH2 PREVLV (PV5) CL IND	MPS E3 LH2 PREVLV (PV6) CL IND	MPS E2 LH2 RECIRC VLV (PV15) CL IND	MPS E3 LH2 RECIRC VLV (PV16) CL IND	MPS E1 LH2 RECIRC VLV (PV14) CL IND	MPS E1 LH2 RECIRC VLV (PV14) OP IND	MPS E2 LH2 RECIRC VLV (PV15) OP IND	MPS E3 LH2 RECIRC VLV (PV16) OP IND	TRS 142 TOPPING VLV (PV13) OP IND	STOP FLOW	A100679 REPL VLV CLOSED #1 IND	AUTO FILL	A100679 REPL VLV OPEN #1 IND	MPS LH2 INBD F/D VLV (PV12) CL IND	A3301 XFR LINE VLV CLOSED #1 IND
CLOSED	OPEN	OPEN	OPEN	OPEN	OPEN	OPEN	NO	NO	NO	CLOSED	CLOSED	CLOSED	CLOSED	CLOSED	CLOSED	OPEN	OPEN	OPEN	OPEN	NO.	CLOSED	NO	OPEN	CLOSED	CLOSED
14243.078	14749.082	14756.078	14764.078	14818.082	14818.082	14819.082	14820.082	14822.082	14824.082	14836.082	14837.082	14839.082	15024.082	15024.082	15041.082	15353.086	15353.086	15353.086	16731.090	18812.102	19245.105	19257.105	20381.109	20406.109	22591.121
261: 1:44:23: 78	261: 1:52:49: 82	261: 1:52:56: 78	261: 1:53: 4: 78	261: 1:53:58: 82	261: 1:53:58: 82	261: 1:53:59: 82	261: 1:54: 0: 82	261: 1:54: 2: 82	261: 1:54: 4: 82	261: 1:54:16: 82	261: 1:54:17: 82	261: 1:54:19: 82	261: 1:57:24: 82	261: 1:57:24: 82	261: 1:57:41: 82	261: 2: 2:53: 85	261: 2: 2:53: 85	261. 2: 2:53: 85	261: 2:25:51: 89	261: 3: 7:39: 0	261: 3: 7:45:105	261: 3:14:50: 0	261: 3:26:41:109	261: 3:27: 6:109	261: 4: 3:31:121

Table 1. STS-35 scrub 3 LH₂ leak isolation timeline (continued)

	** ***		ER THO MENT WILL #1 OBEN TND	
4: 3:55:441 4:33:44: D	22611.441 23508 129	ON EN	ET LHZ VENT VLV #1 OFEN IND STOP FLOW	
:33:50:13	4410.13	CLOSED	A100679 REPL VLV CLOSED #1 IND	
4:37:36: 0	24413.133	NO	AUTO DRAIN	
: 4:40:58:136	24838.137	CLOSED	MPS LH2 HI PT BL VLV (PV22) CL IND	
4:45:29:136	25109.137	OPEN	MPS LH2 HI PT BL VLV (PV22) OP IND	
4:56:20:140	25760.141	CLOSED	MPS LH2 TOPPING VLV (PV13) CL IND	
4:56:33:140	25773.141	OPEN	A100677 MAIN FILL VALVE OPEN IND	
: 4:57: 1:140	25801.141	OPEN	MPS LH2 INBD F/D VLV (PV12) OP IND	
: 5: 0:38:140	26018.141	OFF	MPS E1 LH2 RECIRC PUMP (PP1) SPEED	ENGINE 1 RECIRC TEST
: 5: 0:46:140	26026.141	OPEN	MPS E1 LH2 PREVLV (PV4) OP IND A	
: 5: 5: 3:144	26283.145	NO	MPS E1 LH2 RECIRC PUMP (PP1) SPEED	
: 5: 5:13:144	26293.145	CLOSED	MPS E1 LH2 PREVLV (PV4) CL IND	
: 5: 6:55:144	26395.145	OFF	MPS E2 LH2 RECIRC PUMP (PP2) SPEED	ENGINE 2 RECIRC TEST
: 5: 7: 4:144	26404.145	OPEN	MPS E2 LH2 PREVLV (PV5) OP IND A	
: 5:12: 2:144	26702.145	NO	MPS E2 LH2 RECIRC PUMP (PP2) SPEED	
: 5:12:11:144	26711.145	CLOSED	MPS E2 LH2 PREVLV (PV5) CL IND	
: 5:15:20:148	26900.148	OFF	MPS E3 LH2 RECIRC PUMP (PP3) SPEED	ENGINE 3 RECIRC TEST
: 5:15:29:148	26909.148	OPEN	MPS E3 LH2 PREVLV (PV6) OP IND A	
: 5:20:48:148	27228.148	OFF	MPS E2 LH2 RECIRC PUMP (PP2) SPEED	
: 5:20:58:148	27238.148	OPEN	MPS E2 LH2 PREVLV (PV5) OP IND A	
: 5:24: 1:148	27421.148	NO	MPS E3 LH2 RECIRC PUMP (PP3) SPEED	
: 5:24:10:148	27430.148	CLOSED	MPS E3 LH2 PREVLV (PV6) CL IND	
: 5:26:58:152	27598.152	NO	MPS E2 LH2 RECIRC PUMP (PP2) SPEED	
: 5:27: 8: 0	27608.000	CLOSED	E2 LH2 BLEED VALVE	

Table 1. STS-35 scrub 3 LH₂ leak isolation timeline (continued)

																						RECIRC RETURN	LOCK-UP TEST	
	CL IND	OP IND A			CL IND	OP		(PP3) SPEED		(PP2) SPEED	OP IND A	(PP2) SPEED	CL IND	OP IND A		PUMP (PP3) SPEED	CL IND	(PV5) OP IND A	PUMP (PP2) SPEED		CLOSED IND	(PP1) SPEED	485	PUMP (PP2) SPEED
E2 LH2 BLEED VALVE	MPS E2 LH2 PREVLV (PV5)	MPS E1 LH2 PREVLV (PV4) OP	E1 LH2 BLEED VALVE	E1 LH2 BLEED VALVE	MPS E1 LH2 PREVLV (PV4)	MPS E3 LH2 PREVLV (PV6)	E3 LH2 BLEED VALVE	MPS E3 LH2 RECIRC PUMP (PP3) SPEED	E3 LH2 BLEED VALVE	MPS E2 LH2 RECIRC PUMP	MPS E2 LH2 PREVLV (PV5) OP IND A	MPS E2 LH2 RECIRC PUMP (PP2) SPEED	MPS E2 LH2 PREVLV (PV5)	MPS E1 LH2 PREVLV (PV4)	MPS E1 LH2 RECIRC PUMP	MPS E3 LH2 RECIRC PUMP	MPS E3 LH2 PREVLV (PV6), CL IND	MPS E2 LH2 PREVLV (PV5)	MPS E2 LH2 RECIRC PUMP	STOP FLOW	A100678 AUX FILL VALVE CLOSED IND	MPS E1 LH2 RECIRC PUMP (PP1) SPEED	MPS El LH2 DOEUTY VIVIAGO CHI LE SMM	LH2 RECIRC
OPEN	CLOSED	OPEN	CLOSED	OPEN	CLOSED	OPEN	CLOSED	OFF	OPEN	OFF	OPEN	NO	CLOSED	OPEN	OFF	NO	CLOSED	OPEN	OFF	NO	CLOSED	NO	CLOSED	NO
27854.000	27863.152	27888.152	27893.000	28091.000	28100.152	28122.152	28125.000	28336.156	28358.000	28365.156	28375.156	28869.156	28878.156	28885.156	28891.156	29074.160	29082.160	29087.160	29094.160	29094.160	29264.160	29479.160	29487.160	29495.160
.: 5:31:14: 0	.: 5:31:23:152	.: 5:31:48:152	.: 5:31:52: 0	.: 5:35:11: 0	.: 5:35:20:152	.: 5:35:42:152	:: 5:35:45: 0	1: 5:39:16:156	1: 5:39:38: 0	1: 5:39:45:156	1: 5:39:55:156	1: 5:48: 9:156	1: 5:48:18:156	1: 5:48:25:156	1: 5:48:31:156	1: 5:51:34:160	: 5:51:42:160	.: 5:51:47:160	.: 5:51:54:160	.: 5:54:32: 0	.: 5:54:44:160	.: 5:58:19:160	.: 5:58:27:160	.: 5:58:35:160
261	261	261	261	261	261	261	261	261	261	261	261	261	261	261	261	261	9	261	261	261	261	261	261	261

Table 1. STS-35 scrub 3 LH₂ leak isolation timeline (continued)

261: 5:58:42:160	29502.160	CLOSED	E2 LH2 PREVLV (PV5) CL IN
261: 6: 0:32:160	29612.160	OFF	MPS E2 LH2 RECIRC PUMP (PP2) SPEED
261: 6: 0:42:160	29622.160	OPEN	MPS E2 LH2 PREVLV (PV5) OP IND A
261: 6: 0:48:160	29628.160	OFF	MPS E1 LH2 RECIRC PUMP (PP1) SPEED
261: 6: 0:55:160	29635.160	OPEN	MPS E1 LH2 PREVLV (PV4) OP IND A
261: 6: 1: 3: 0	29643.000	CLOSED	E1 LH2 BLEED VALVE
(61: 6: 1:10: 0	29650.000	CLOSED	E2 LH2 BLEED VALVE
261: 6: 1:14:160	29654.160	CLOSED	MPS LH2 4IN DISC VLV (PD3) CL IND
261: 6: 1:24:160	29664.160	OFF	MPS E3 LH2 RECIRC PUMP (PP3) SPEED
261: 6: 1:25:160	29665.160	NO	MPS E3 LH2 RECIRC PUMP (PP3) SPEED
261: 6: 1:27:160	29667.160	OFF	MPS E3 LH2 RECIRC PUMP (PP3) SPEED
261: 6: 1:32:160	29672.160	CLOSED	MPS E2 LH2 RECIRC VLV (PV15) CL IND
261: 6: 1:32:160	29672.160	CLOSED	MPS E3 LH2 RECIRC VLV (PV16) CL IND
261: 6: 2:19:160	29719.160	OPEN	MPS E3 LH2 PREVLV (PV6) OP IND A
261: 6: 5:29:164	29909.164	CLOSED	MPS E1 LH2 PREVLV (PV4) CL IND
261: 6: 5:53:164	29933.164	OPEN	MPS E1 LH2 PREVLV (PV4) OP IND A
261: 6: 6:57:164	29997.164	OPEN	MPS LH2 TOPPING VLV (PV13) OP IND
261: 6: 7: 6: 0	30007.000	OPEN	E1 LH2 BLEED VALVE
261: 6: 7:14: 0	30014.000	OPEN	E2 LH2 BLEED VALVE
261: 6: 7:39:164	30039.164	CLOSED	MPS E1 LH2 PREVLV (PV4) CL IND
261: 6: 7:43:164	30043.164	CLOSED	MPS E2 LH2 PREVLV (PV5) CL IND
261: 6: 7:46:164	30046.164	CLOSED	MPS E3 LH2 PREVLV (PV6) CL IND
261: 6: 7:55:164	30055.164	CLOSED	MPS LH2 HI PT BL VLV (PV22) CL IND
261: 6:10:24: 0	30055.164	NO	AUTO DRAIN
261: 6:11:39:164	30279.164	CLOSED	MPS E1 LH2 RECIRC VLV(PV14) CL IND
261: 6:12:41:164	30341.164	CLOSED	MPS LH2 TOPPING VLV (PV13) CL IND

Table 1. STS-35 scrub 3 LH₂ leak isolation timeline (continued)

A100677 MAIN FILL VALVE OPEN IND	MPS LH2 TOPPING VLV (PV13) OP IND
OPEN	OPEN
30355.164	30407.164
261: 6:12:55:164	261: 6:13:47:164

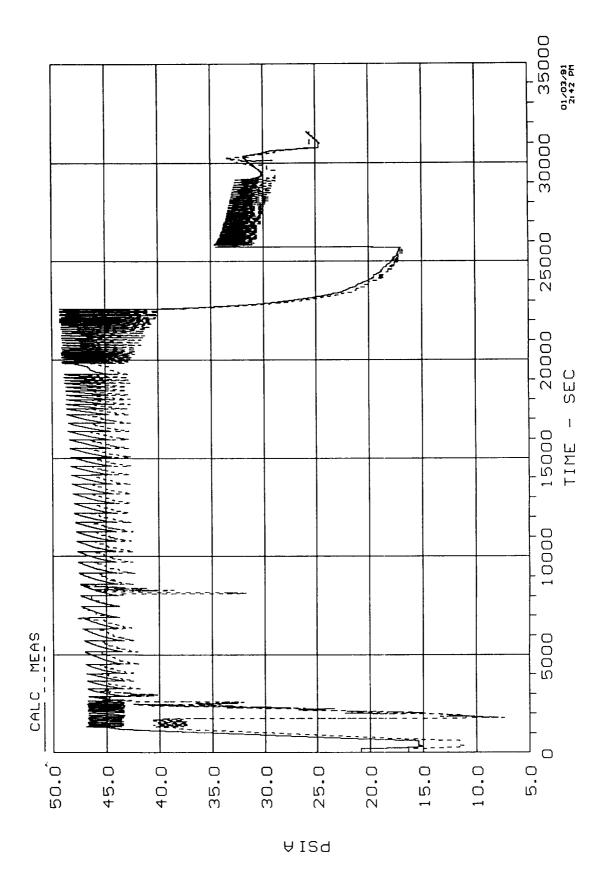


Figure 1. STS-35 S3 manifold pressure.

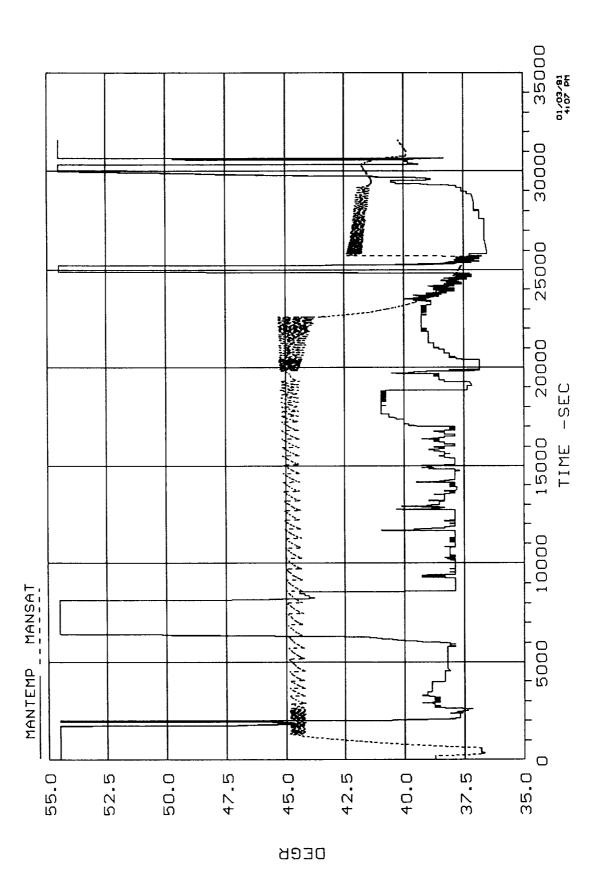


Figure 2. STS-35 S3 manifold temperature

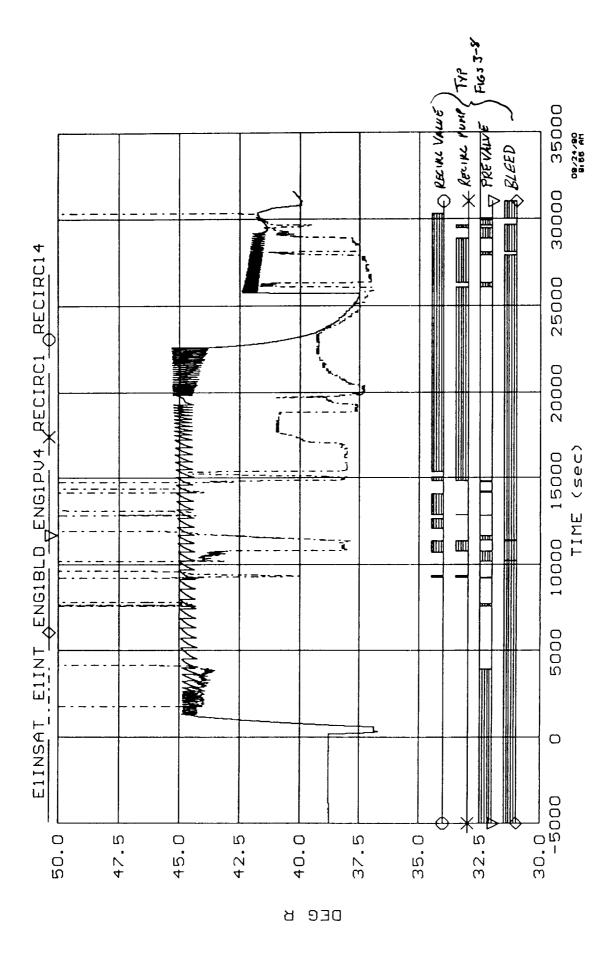


Figure 3. STS-35 S3 engine 1 inlet temperature.

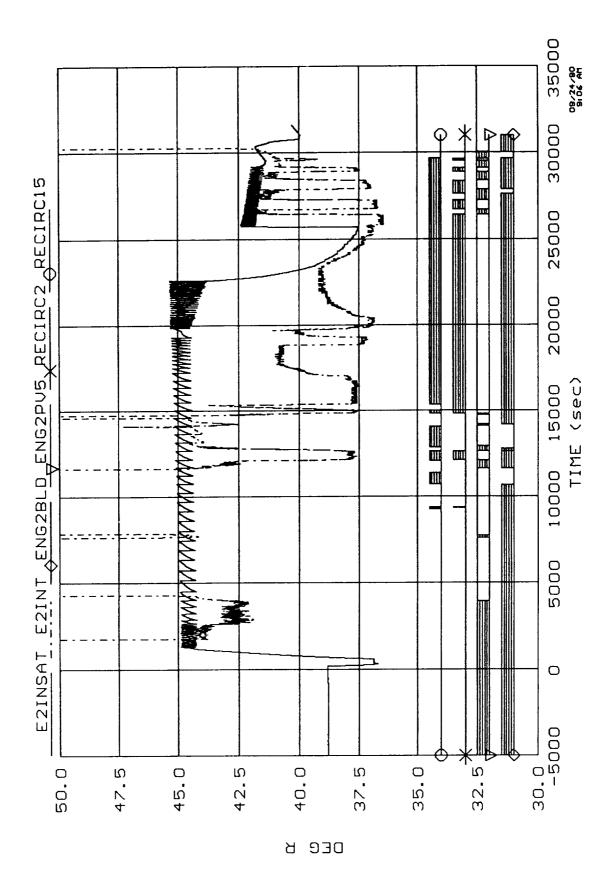


Figure 4. STS-35 S3 engine 2 inlet temperature.

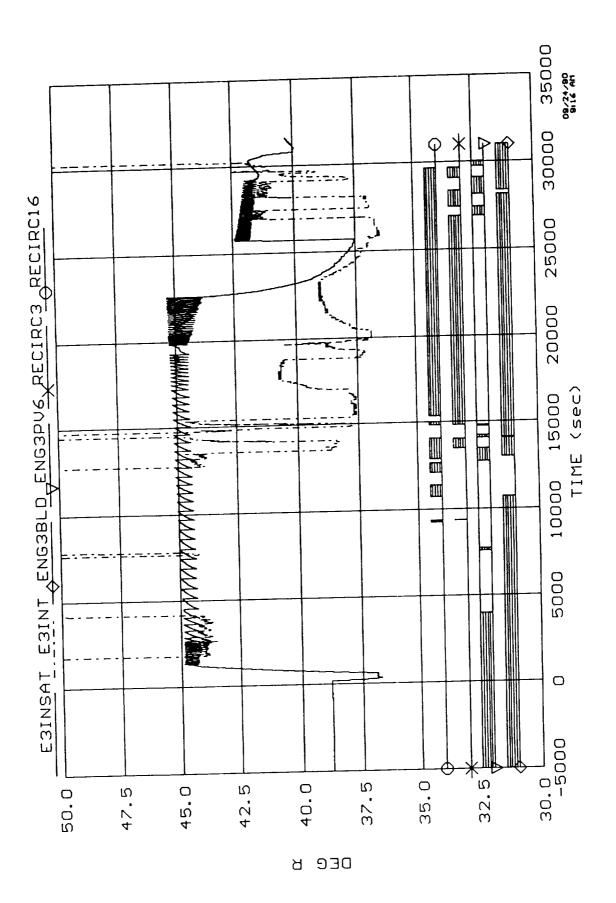


Figure 5. STS-35 S3 engine 3 inlet temperature.

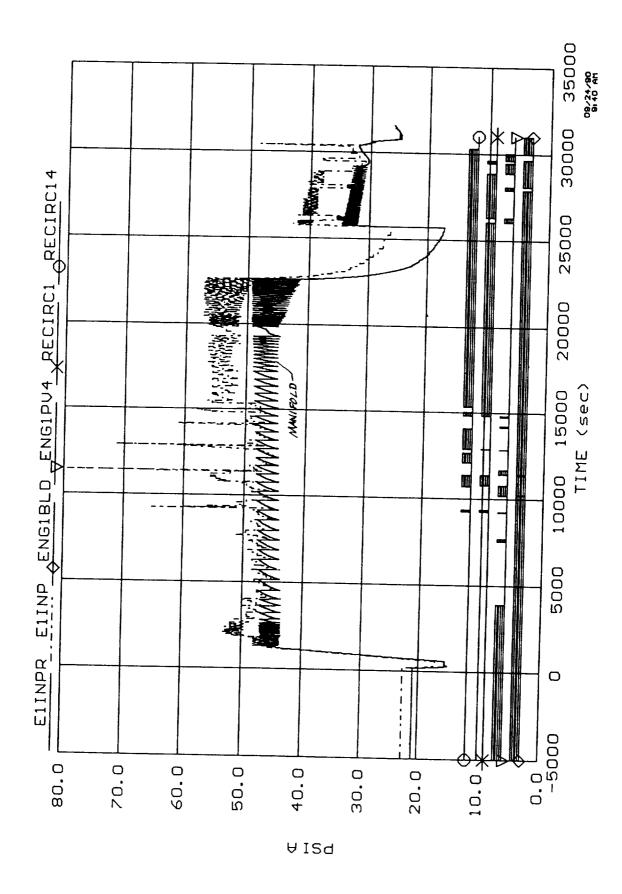


Figure 6. STS-35 S3 engine 1 inlet pressure.

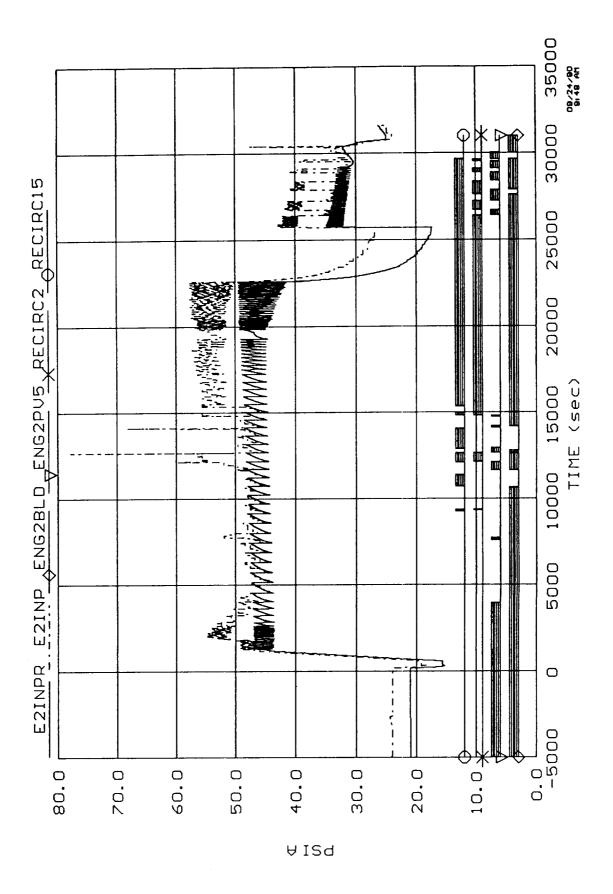


Figure 7. STS-35 S3 engine 2 inlet pressure.

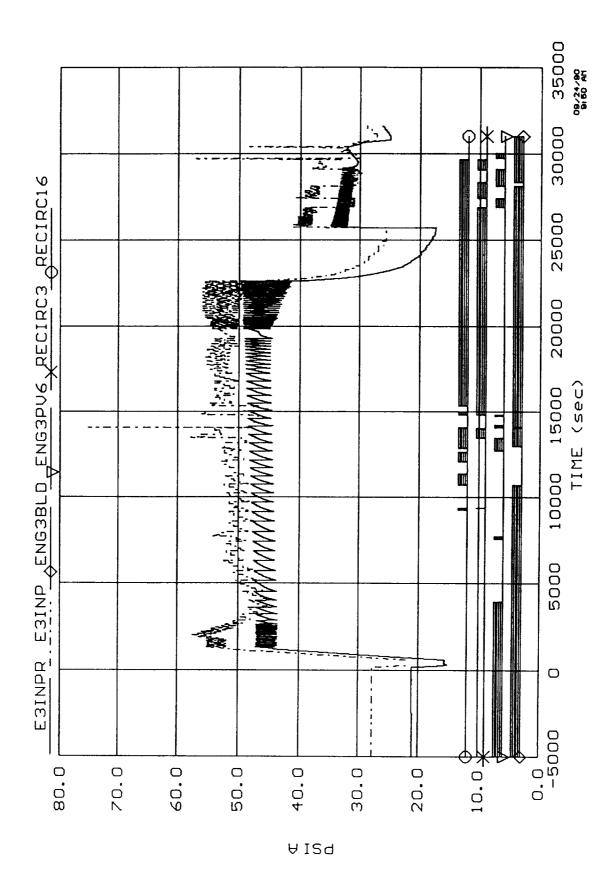


Figure 8. STS-35 S3 engine 3 inlet pressure.

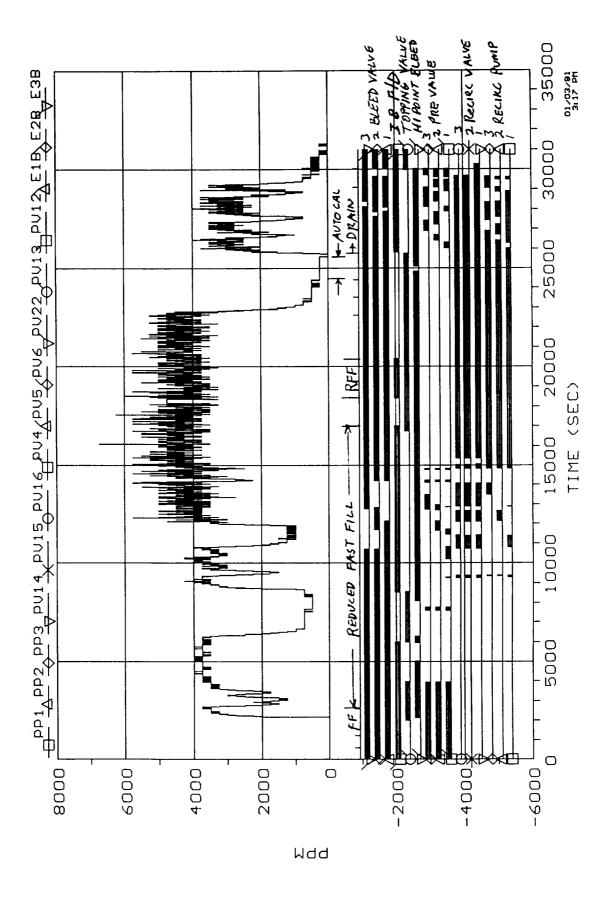


Figure 9. STS-35 S3 aft compartment high range H2 concentration.

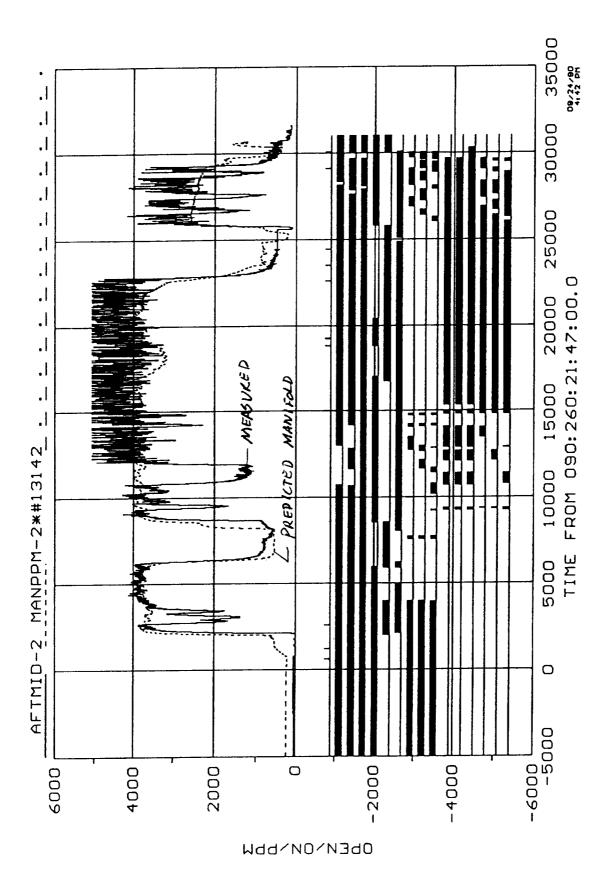


Figure 10. STS-35 scrub 3.

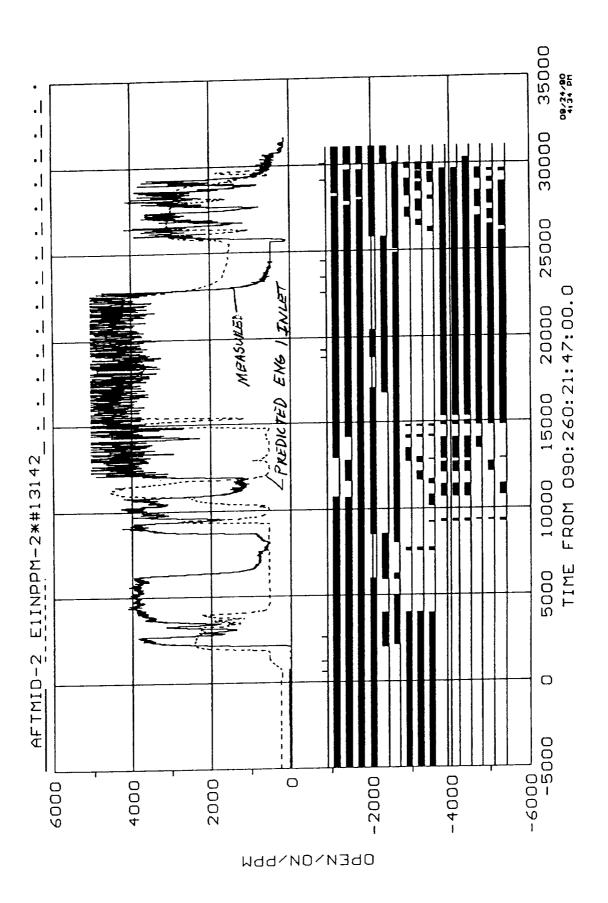


Figure 11. STS-35 scrub 3.

Figure 12. STS-35 scrub 3.

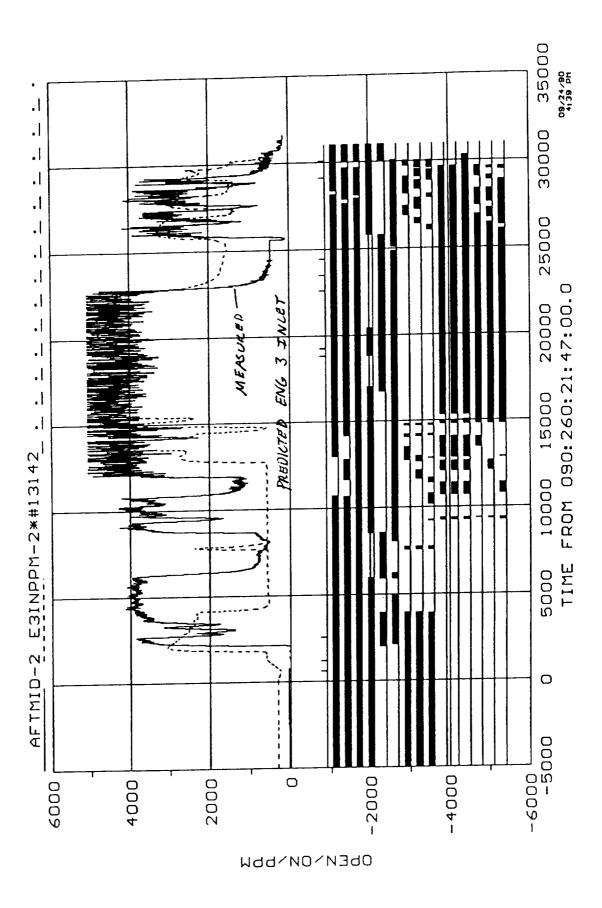


Figure 13. STS-35 scrub 3.

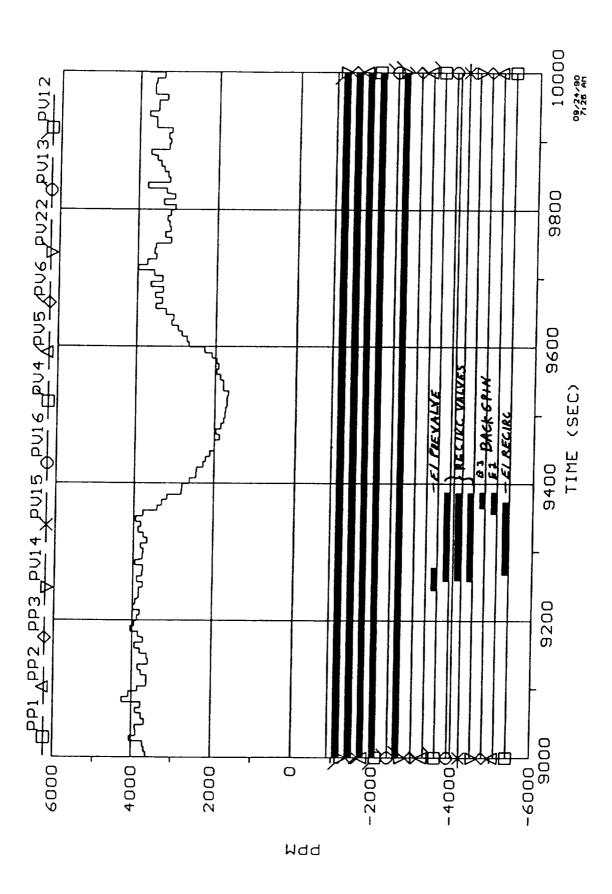


Figure 14. STS-35 S3 aft compartment H2 concentration.

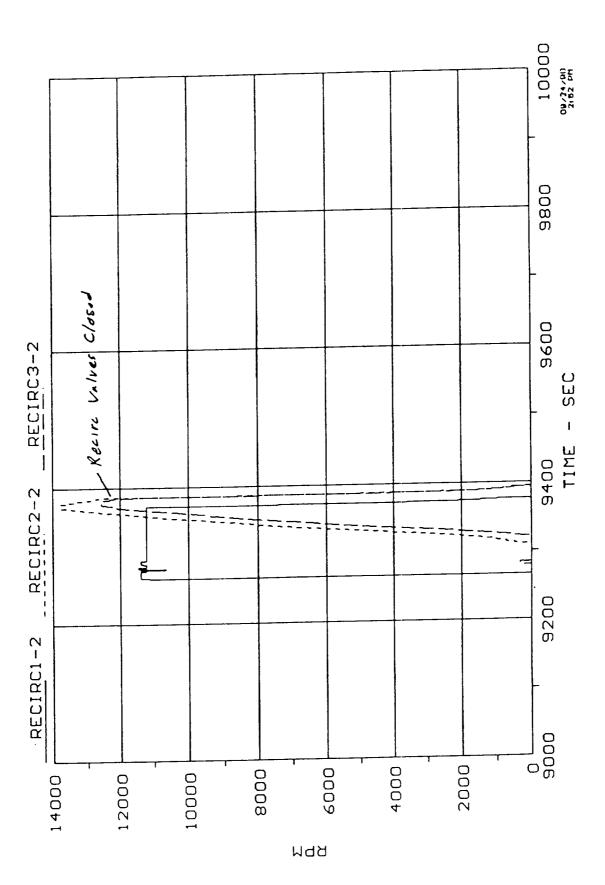


Figure 15. STS-35 S3 recirculation pump speed.

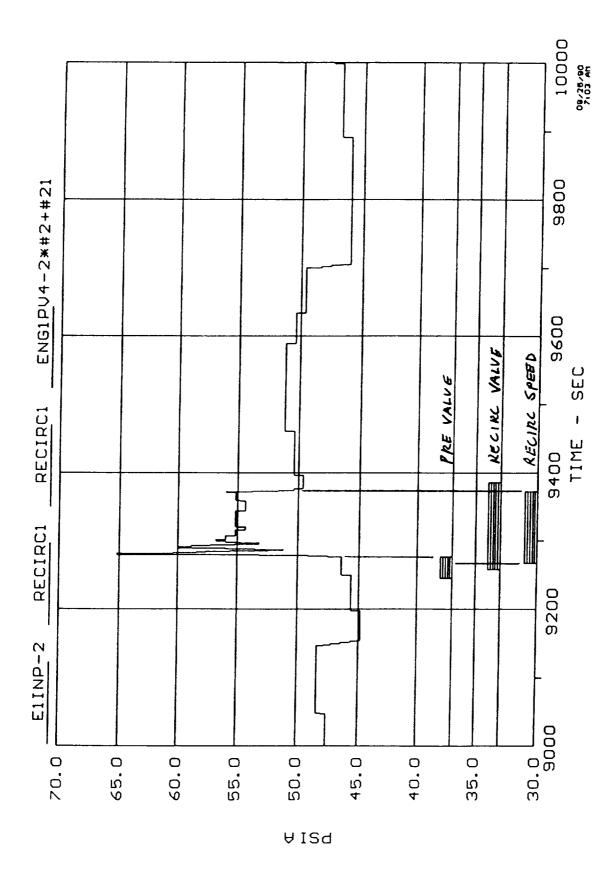


Figure 16. STS-35 S3 engine 1 inlet pressure.

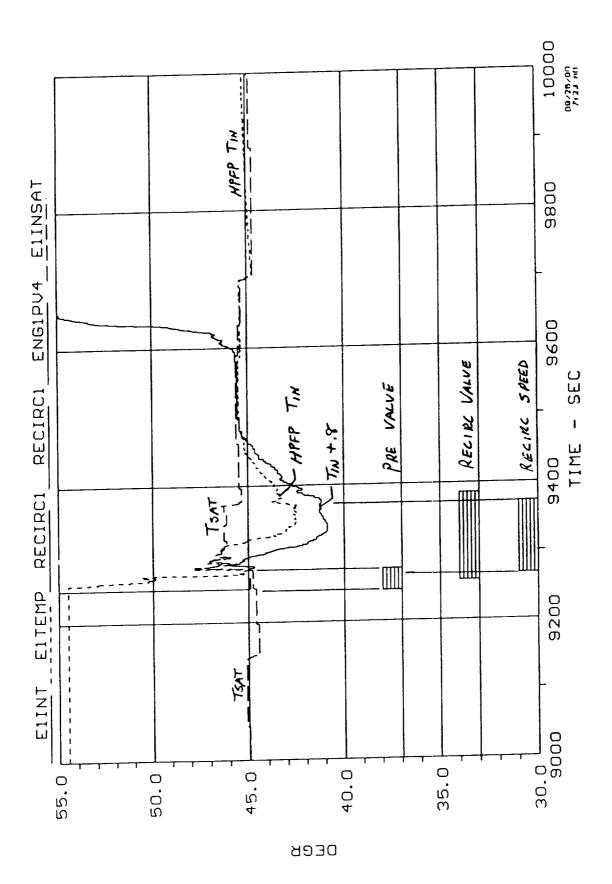


Figure 17. STS-35 S3 engine 1 inlet temperature.

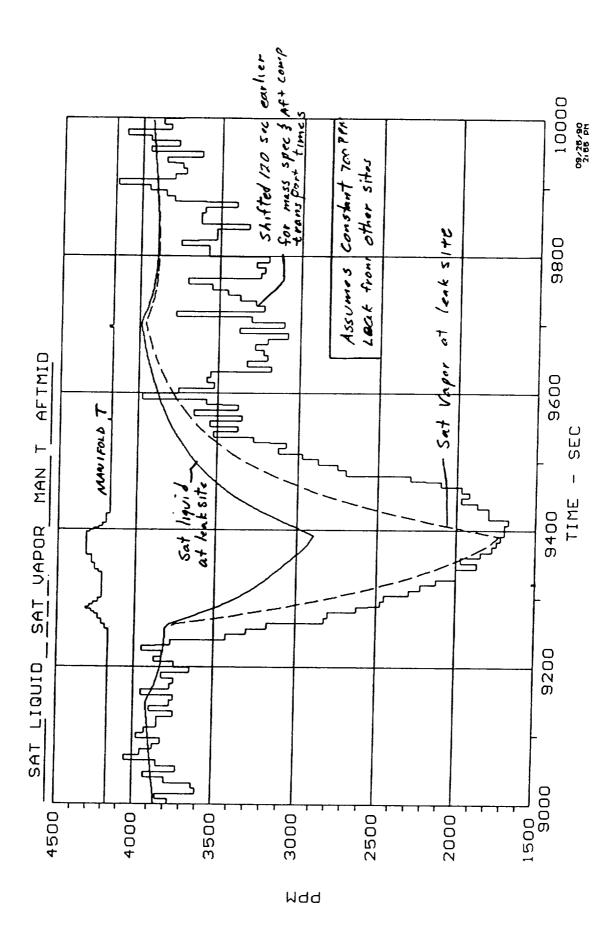


Figure 18. STS-35 S3 predicted aft compartment H2 concentration.

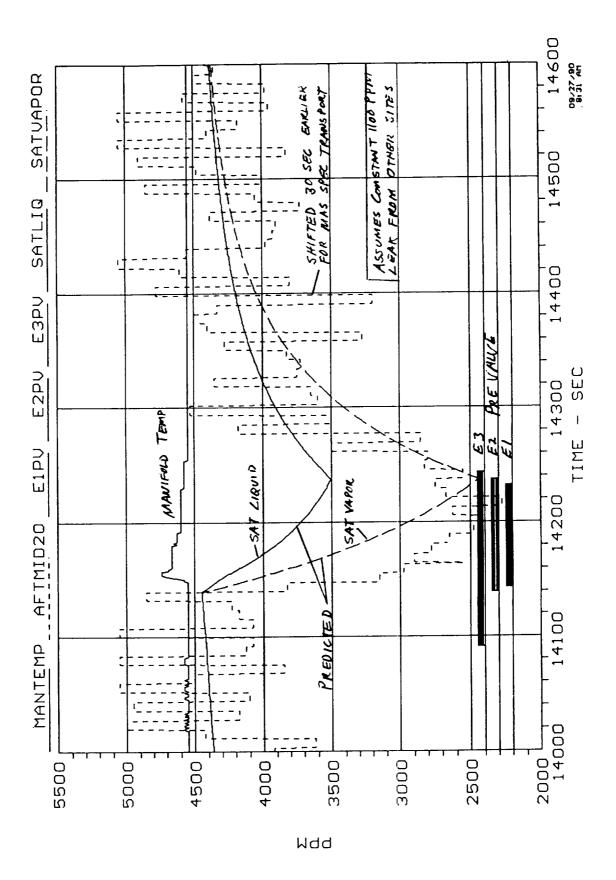


Figure 19. STS-35 S3 predicted aft compartment H₂ concentration.

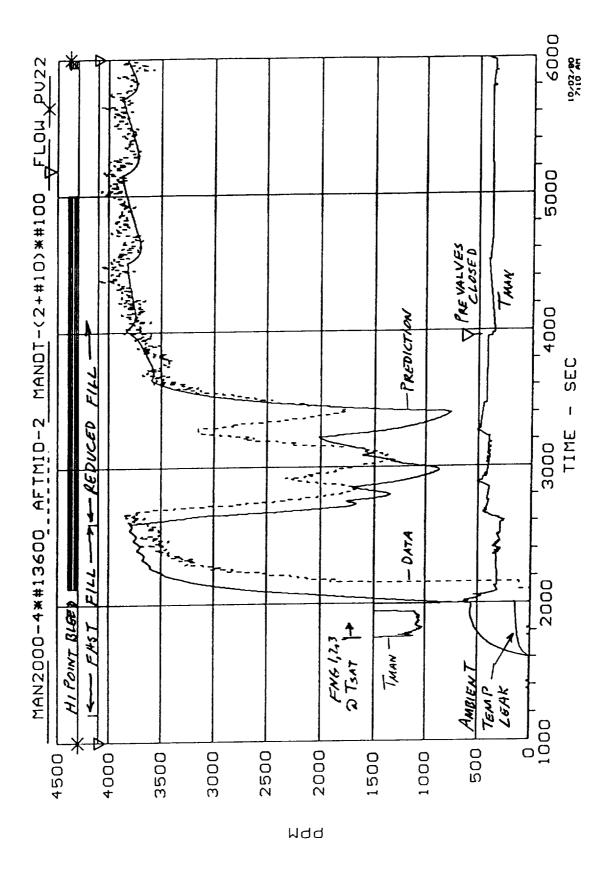


Figure 20. STS-35 S3 fast-fill/reduced-flow transition.

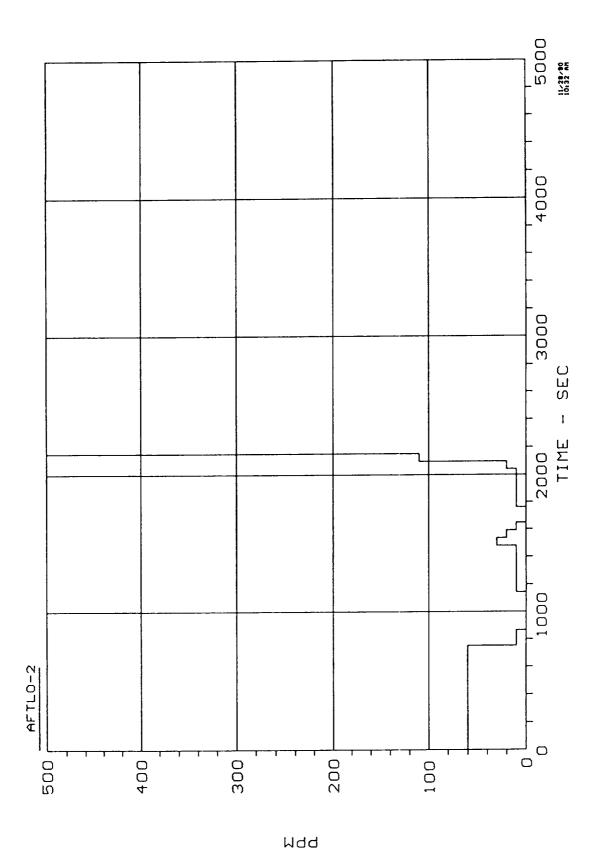


Figure 21. STS-35 S3 low range aft compartment H2 concentration.

Figure 22. STS-35 S3 manifold flow.

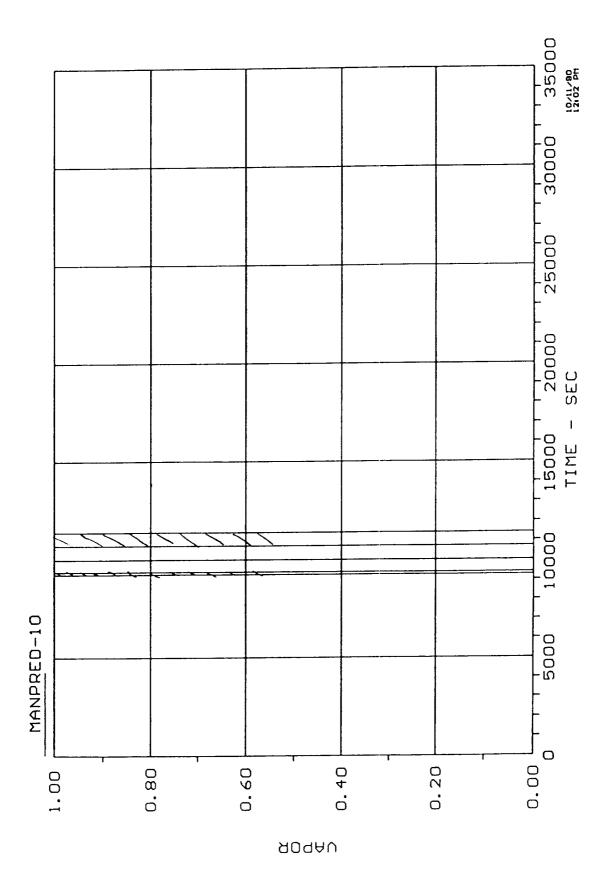


Figure 23. STS-35 S3 engine 2 recirculation pump vapor.

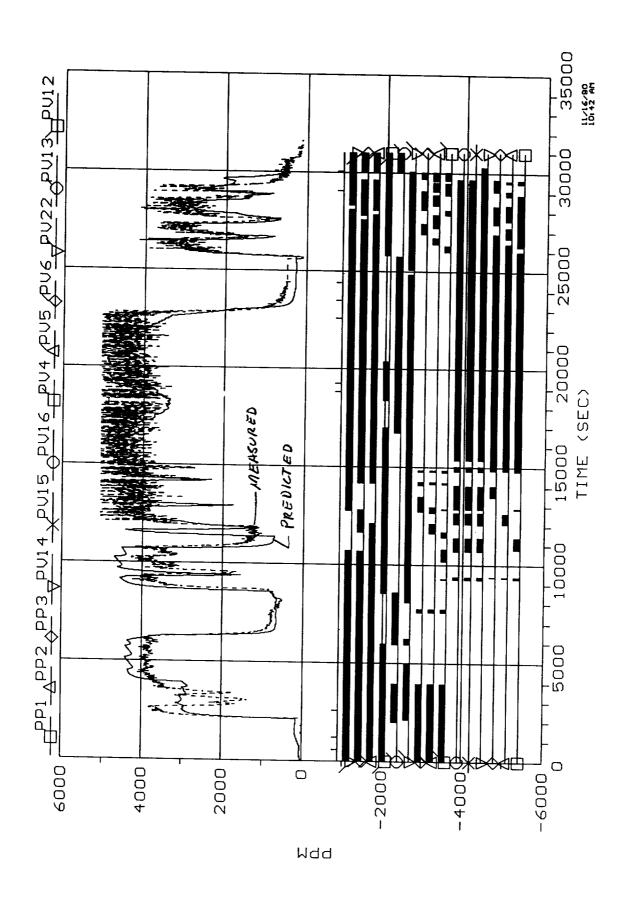


Figure 24. STS-35 S3 predicted PV5 leak.

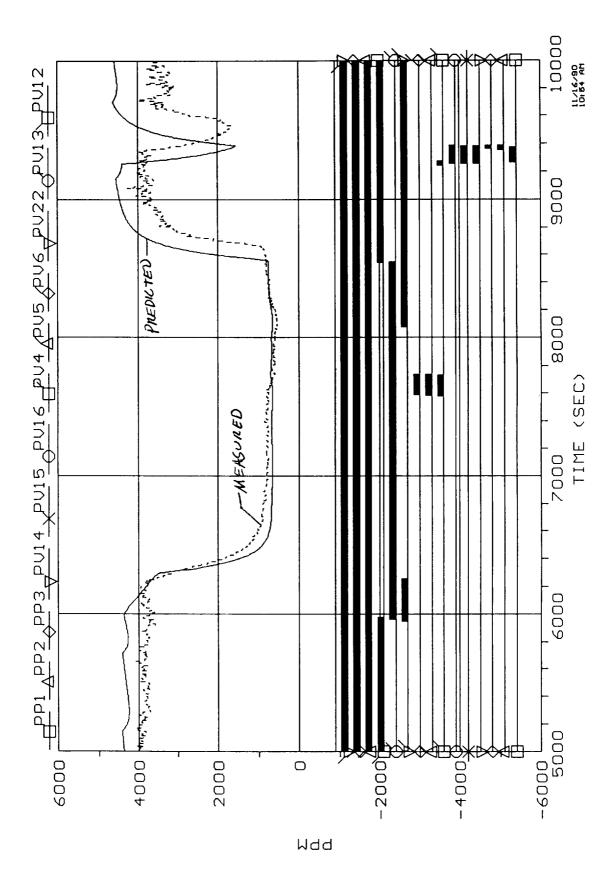


Figure 25. STS-35 S3 predicted PV5 leak.

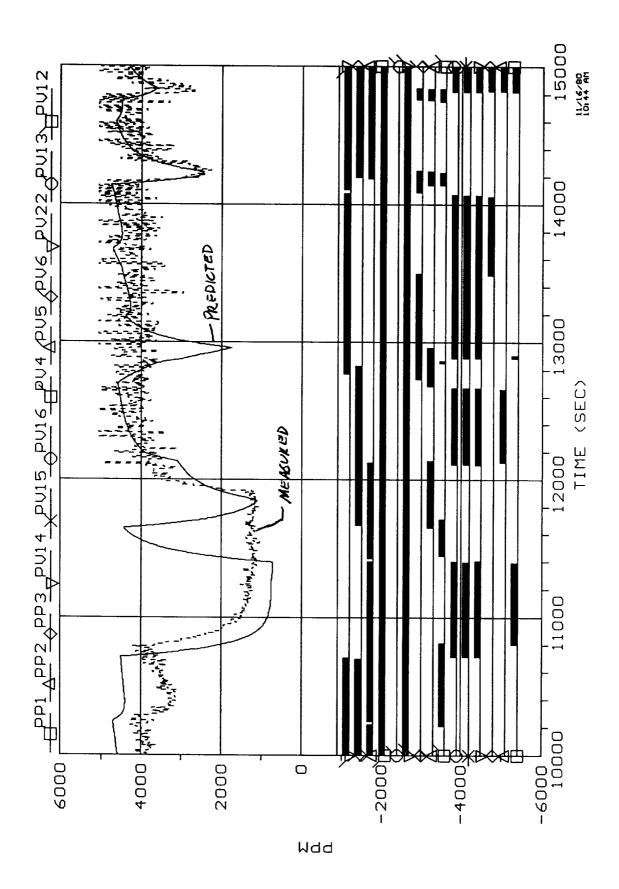


Figure 26. STS-35 S3 predicted PV5 leak.

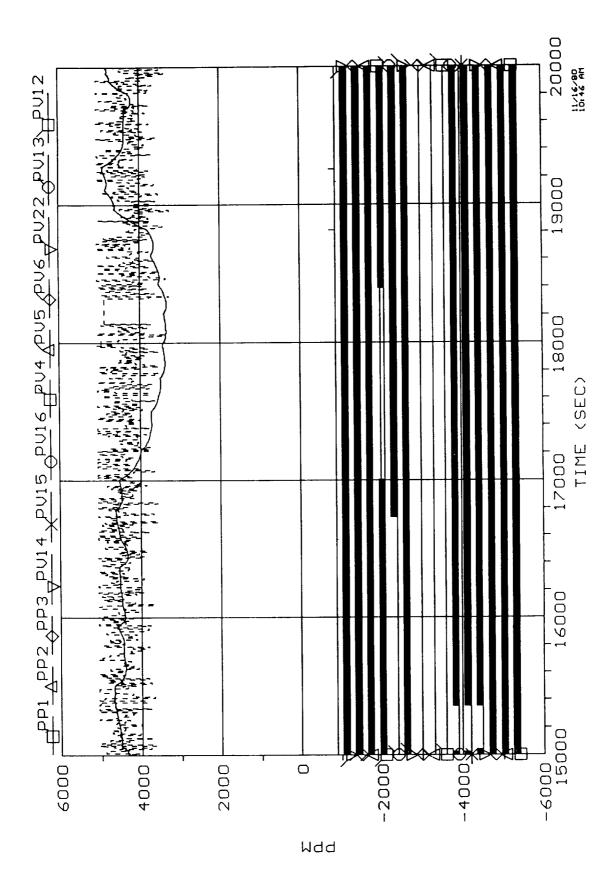


Figure 27. STS-35 S3 predicted PV5 leak.

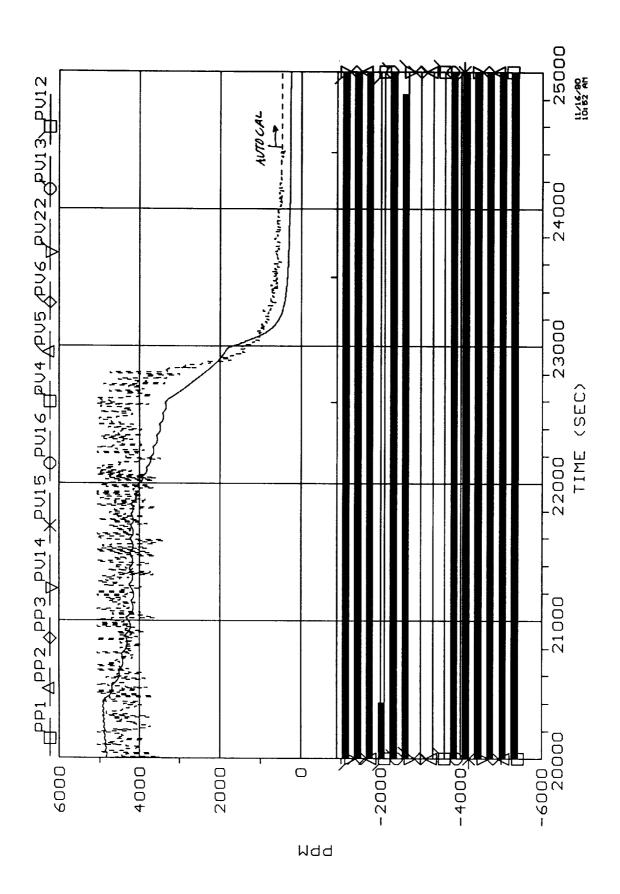


Figure 28. STS-35 S3 predicted PV5 leak.

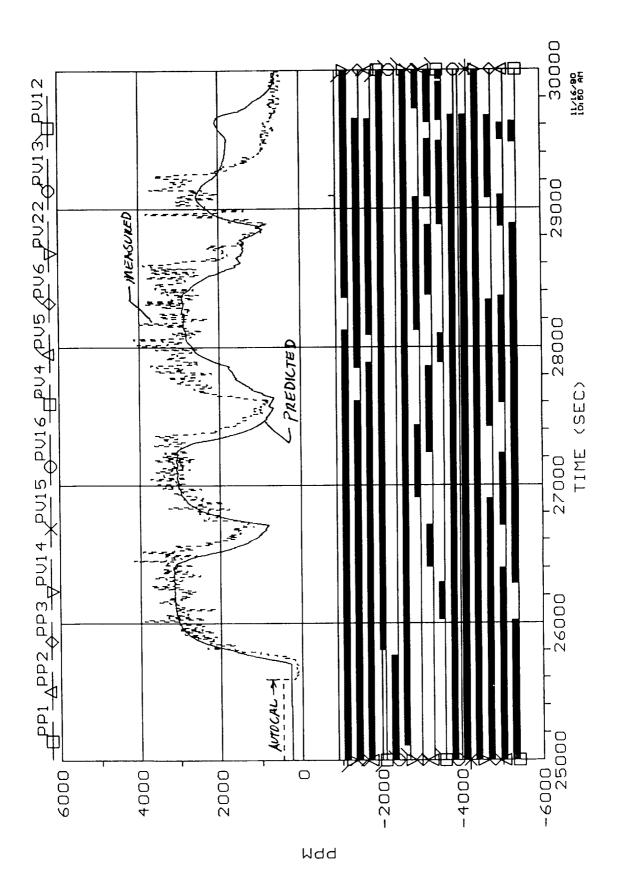


Figure 29. STS-35 S3 predicted PV5 leak.

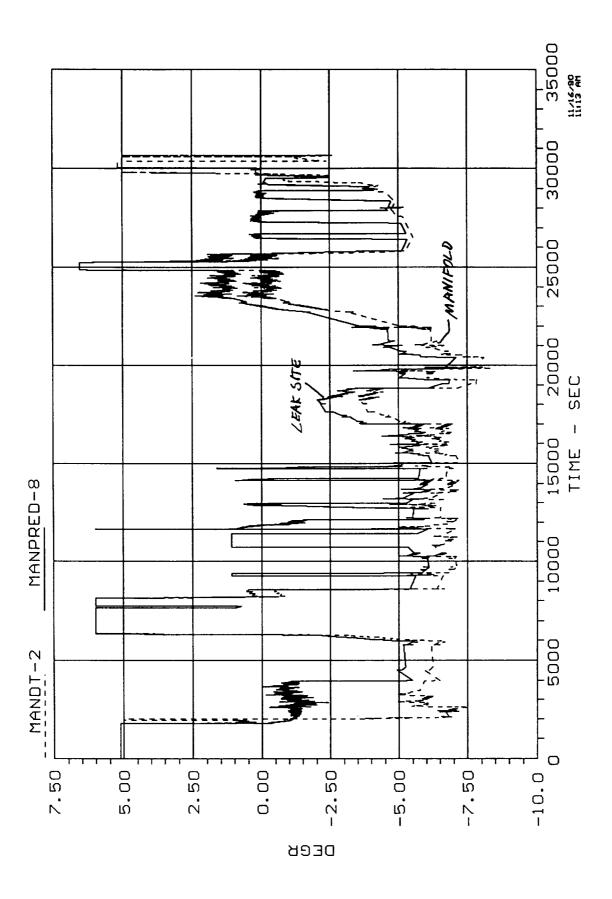


Figure 30. STS-35 S3 leak site subcool temperature.

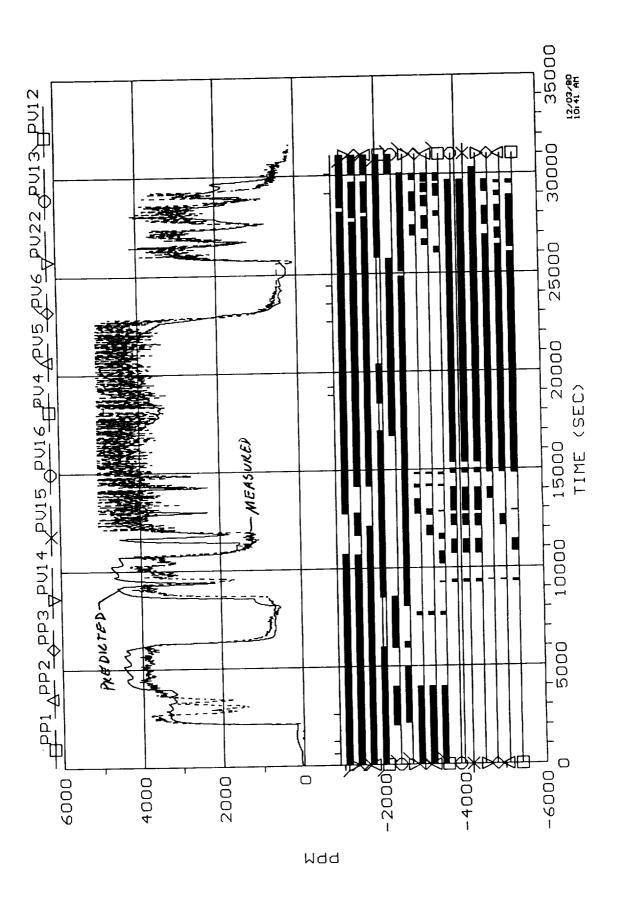


Figure 31. STS-35 S3 80-percent PV5 + 20-percent 17-in line leak.

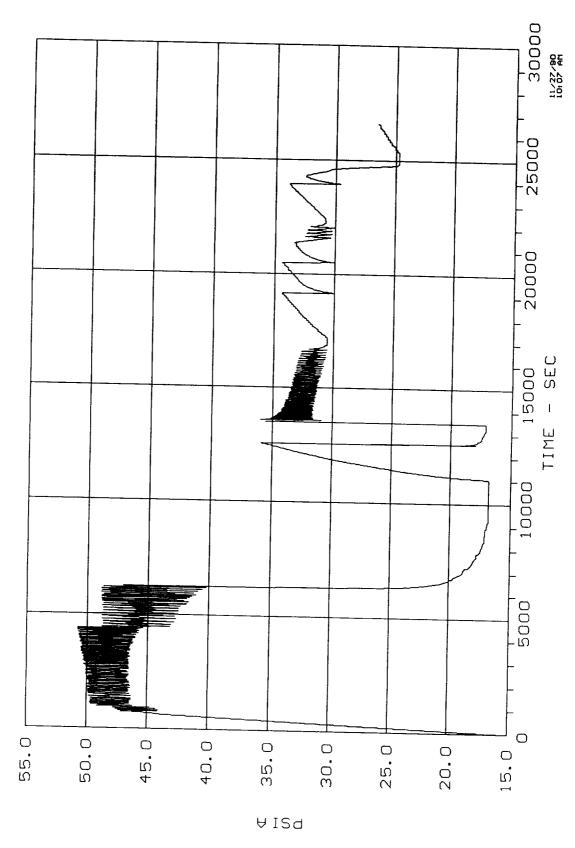


Figure 32. STS-35 S2 manifold pressure.

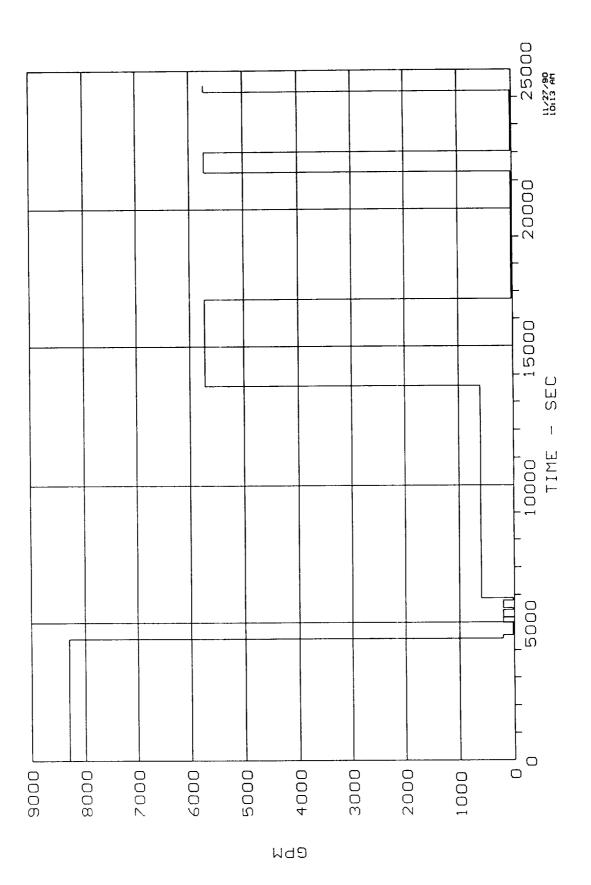


Figure 33. STS-35 S2 flow.

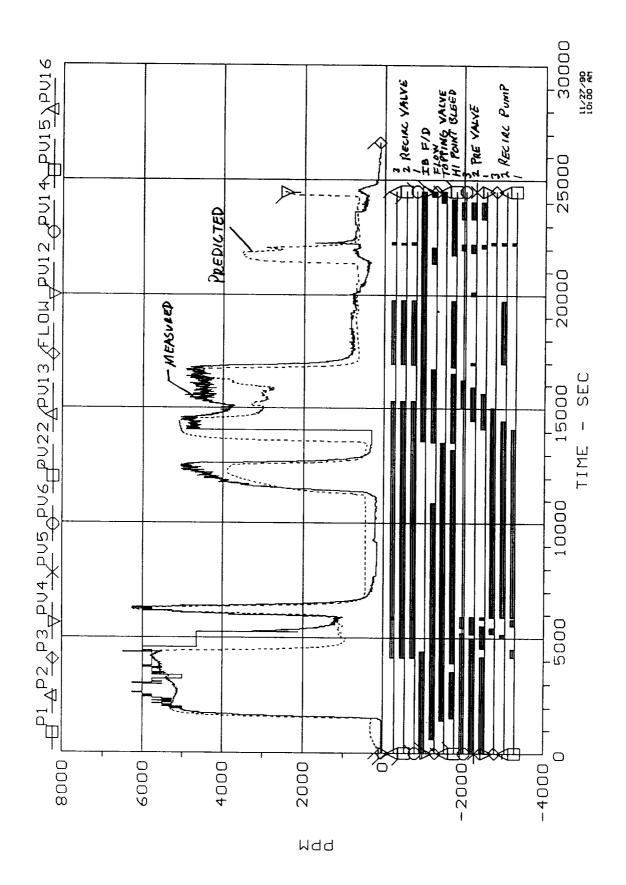


Figure 34. STS-35 S2 predicted versus actual aft H2 concentration.

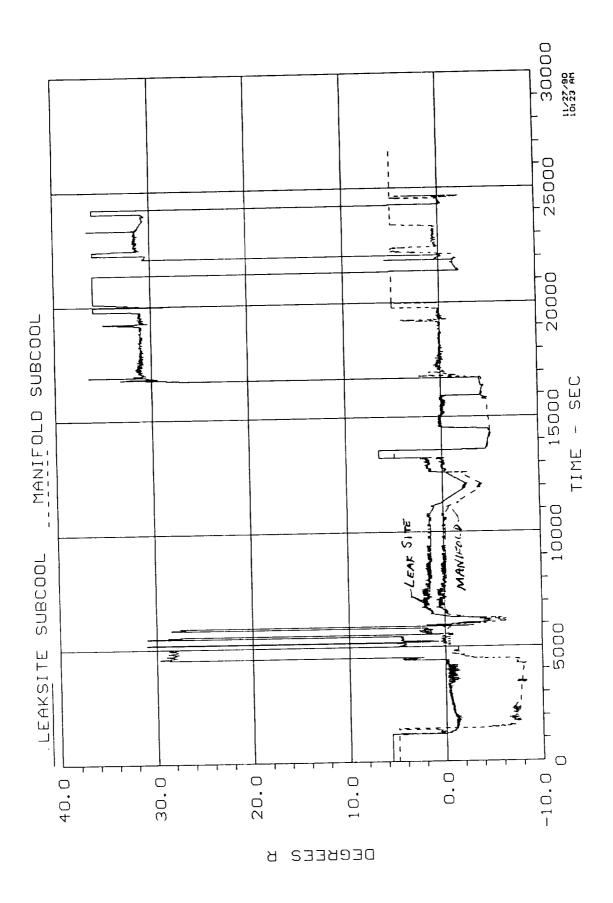


Figure 35. STS-35 S2 leak site subcool versus manifold subcool.

Figure 36. STS-35 T1 manifold pressure.

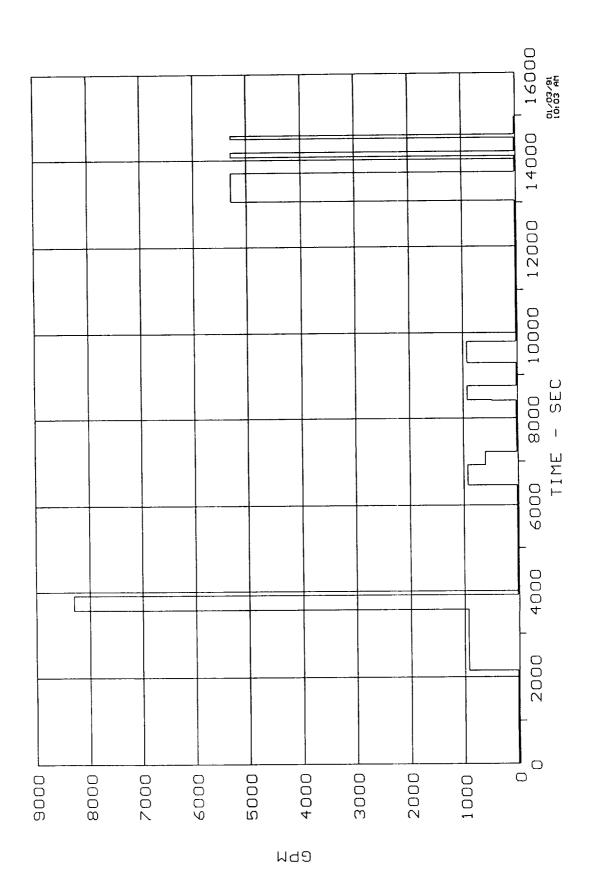


Figure 37. STS-35 T1 flow.

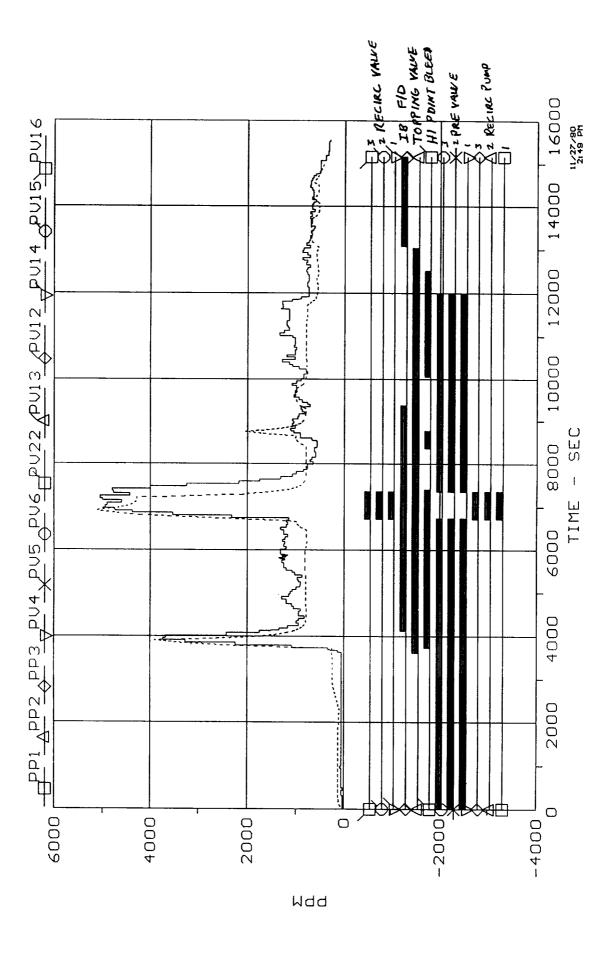


Figure 38. STS-35 T1 predicted versus actual aft H2 concentration.

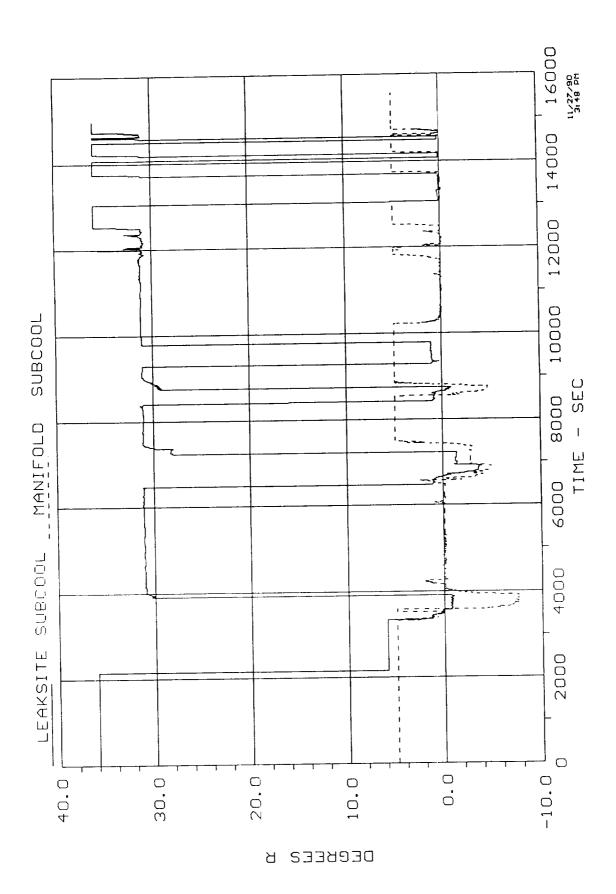


Figure 39. STS-35 T1 leak site subcool versus manifold subcool.

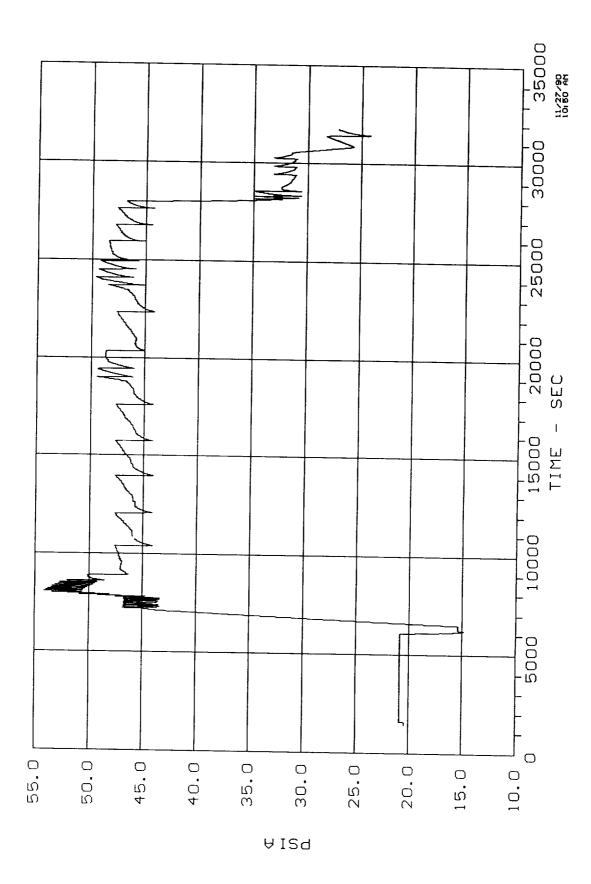


Figure 40. STS-35 S1 manifold pressure.

Figure 41. STS-35 S1 flow.

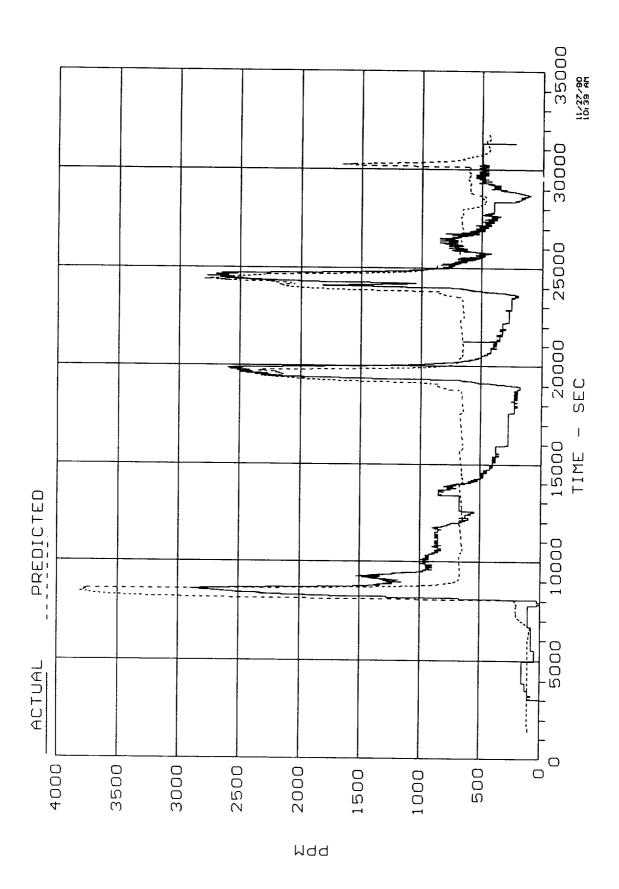


Figure 42. STS-35 S1 predicted versus actual aft H₂ concentration.

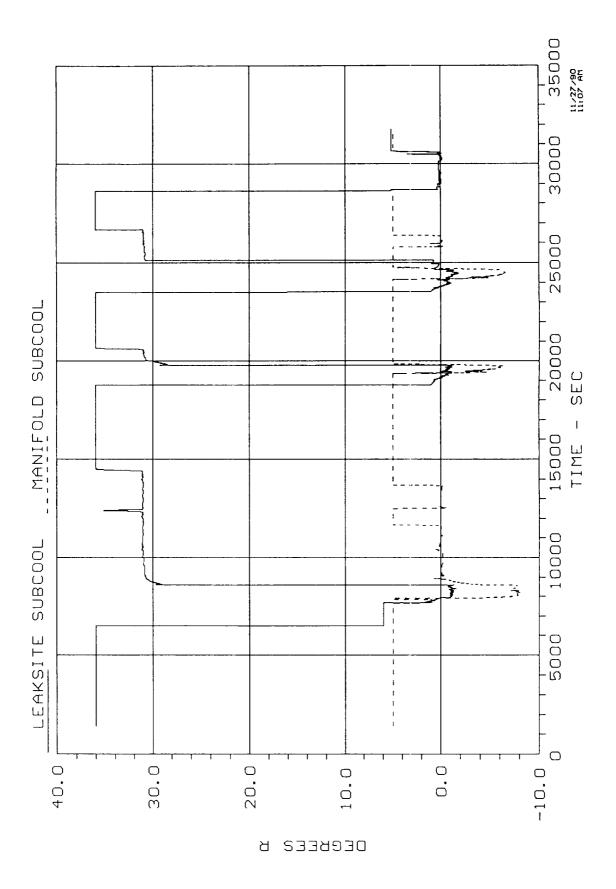


Figure 43. STS-35 S1 leak site subcool versus manifold subcool.

		<u></u>

APPENDIX

```
c Routine to calculate Aft Comp PPM based on subcooling and pressure
c STS35S3 PV5 leak model
C
        parameter( vol
                        = 4550.
                                              ! aft comp volume (ft3)
        parameter( volin3= vol*1728.)
                                              ļ
                                                  11
                                                                 (in3)
        parameter( purge = 235. )
                                              ! aft comp purge
                                                                (lb/min)
        parameter( w
                        = 1728.*purge/.072 )! purge flow
                                                                 (scim)
C
        parameter( nrec1 = 2 )
                                 !recirc1
        parameter( nrec2 = 3 )
                                 !recirc2
        parameter( nrec3 = 4 )
                                 !recirc3
        parameter(npv14 = 5)
                                 !pv14
        parameter( npv15 = 6 )
                                 !pv15
        parameter( npv16 = 7 )
                                 !pv16
        parameter( npv4 = 8 )
                                 !pv4
        parameter(npv5 = 9)
                                 !pv5
        parameter( npv6 = 10)
                                 !pv6
        parameter(npv22 = 11)
                                 !pv22
        parameter(npv13 = 12)
                                 !pv13
        parameter( npv12 = 13)
                                 !pv12
        parameter( nbld1 = 14)
                                !elbld
        parameter( nbld2 = 15)
                                !e2bld
        parameter( nbld3 = 16)
                                !e3bld
C
        real a1(34)
        real a2(22)
        real event(17)
        real tempin(3)
С
C
        LEAK RATE TABLE
С
        data a1/32...0., -15.35, 2.16, -11.35, 2.12, -7.35, 1.82,
           -5.35, 1.62, -4.35, 1.5, -3.35, 1.37, -2.35, 1.24,
     *
           -1.35, 1.14, -.35, 1.03, 0.0, 1.0, .1, .28, .5, .274,
     *
           .65, .27, 4.65, .25, 14.65, .22, 24.65, .21/
С
        data a2/20.0, 0., 20., .35, 25., .50, 30., .60,
     *
            35., .72, 40., .81, 45., .91, 50., 1.0,
            55., 1.09, 60., 1.18, 70., 1.28 /
C
        TSAT(psia) = 9.969152*(psia)**.2843842 + 15.07444
C
C
        open(unit=1, file='d:\sts35s3\mandt.bin', status='old',
            form='unformatted',access='direct')
C
        open(unit=2, file='d:\sts35s3\manpr.cal', status='old',
            form='unformatted',access='direct')
C
        open(unit=3, file='d:\sts35s3\manpred.bin', form='unformatted',
     *
            status='unknown', access='direct', recl=40)
C
        open(unit=7,file='d:\sts35s3\t\events.evn',status='old')
```

```
C
        open(unit=8,file='d:\sts35s3\elindt.bin',status='old',
            form='unformatted',access='direct')
        open(unit=9,file='d:\sts35s3\e2indt.bin',status='old',
            form='unformatted',access='direct')
        open(unit=10, file='d:\sts35s3\e3indt.bin', status='old',
            form='unformatted',access='direct')
C
        write(5,'(a)') ' Enter Tsat Tolerance '
        read(5,*) tolinp
C
        read(1 ) timb, tempb
        read(2) tim2,prs1
        read(8) time1, teme1
C
         read(9 ) time2, teme2
         read(10) time3, teme3
C
C
               = tim2
         tim
         timxv = tim
         tprint= tim
         prs2 = prs1
         ppm1 = 0.
         ppm2 = 0.
         ppm3 = 0.
C
         MAIN TIME LOOP
C
         continue
   200
                                            ! Time step fixed at 1 sec
         tim=tim+1.
         tol = tolinp
 С
         SET MANIFLOD FLOW RATE
 C
 C
         if( tim .lt. 2590 ) then
            flow = 8300.
         else if( tim .lt. 17004. ) then
            flow = 930.
         else if( tim .lt. 18392. ) then
            flow = 600.
          else if( tim .lt. 20406. ) then
            flow = 930.
          else if (tim .lt. 25801.) then
            flow = 600.
          else if( tim .lt. 35000. ) then
            flow = 5700.
          end if
          continue
    205
          if( tim .lt. timb ) goto 210
            tima=timb
            tempa=tempb
                                                  ! Update manifold temp
            read(1,end=900) timb,tempb
            tprint=tim-1.
          goto 205
```

```
C
  210
        continue
        if( timb-tima .gt. .01 )
          tempm=tempa+(tim-tima)*(tempb-tempa)/(timb-tima)
C
  215
        continue
C
        if (tim .lt. time1) goto 216
          time1p=time1
С
C
          temelp=temel
С
          read(8,end=900) time1,teme1
                                               ! Update engine 1 temp
C
          tprint=tim-1.
C
        goto 215
C
  216
        continue
C
        if( time1-time1p .gt. .01 )
          tempin(1) = teme1p + (tim-time1p) * (teme1-teme1p) / (time1-time1p)
C
  315
        continue
        if( tim .lt. time2 ) goto 316
          time2p=time2
          teme2p=teme2
          read(9,end=900) time2,teme2
                                           ! Update engine 2 temp
          tprint=tim-1.
        goto 315
C
  316
        continue
        if( time2-time2p .gt. .01 )
     1
          tempin(2) = teme2p + (tim-time2p) * (teme2-teme2p) / (time2-time2p)
C
  415
        continue
        if( tim .lt. time3 ) goto 416
С
          time3p=time3
C
          teme3p=teme3
C
          read(10,end=900) time3,teme3
                                           ! Update engine 3 temp
C
          tprint=tim-1.
C
        goto 415
C
  416
        continue
        if( time3-time3p .gt. .01 )
C
          tempin(3)=teme3p+(tim-time3p)*(teme3-teme3p)/(time3-time3p)
C
     1
C
  218
        continue
        if( tim .lt. timxv ) goto 219
          read(7,fmt=*,end=900) timxv,event
                                               ! Update events
          read(7,fmt=*,end=900) timxv
                                              ! Look ahead for next time
          backspace (7)
          tprint=tim-1.
        goto 218
C
  219
        continue
C
220
        if(tim.lT.tim2) goto 230
        tim1=tim2
        prs1=prs2
```

```
! Update pressure
        read(2,end=900) tim2,prs2
        tprint=tim-1.
        goto 220
C
        continue
 230
        if( tim2-tim1 .gt. .01 )
          prs=prs1+(tim-tim1)*(prs2-prs1)/(tim2-tim1)
     1
C
C
        CHECK FOR VAPOR AT PV5 INLET FROM ENG 2 RECIRC PUMP INLET
C
C
        vapor = 0.0
        if (event(npv15) .gt. .1 .and.
               event(nrec2) .lt. .1 .and.
     1
               event(npv5) .lt. .1 .and.
     2
                                            ) vapor = vapor+1.
               tempin(2) .gt. 1.
     3
C
        SET LEAK TEMPERATURE
C
C
         if( event(npv5).gt. .1 ) then
           temp = tempin(2)
         else
           if( vapor .gt. .1 ) then
                                      ! Assume sat vapor leak
             temp = .1
           else
             temp = tempm
           end if
         end if
C
         temp = temp+1.*930./flow
C
         GET LEAK FLOW
C
 C
         call intrp(a1,temp-tol,fact1a)
         call intrp(a1,temp,fact1b)
         call intrp(a1,temp+tol,fact1c)
         call intrp(a1, tempm, factlm)
 C
         call intrp(a2,prs,fact2)
 C
         wleak1 = fact1a*fact2
         wleak2 = fact1b*fact2
         80% PV5, 20% manifold case
 C
         wleak2 = .8*fact1b*fact2+.2*fact1m*fact2
         wleak3 = fact1c*fact2
 C
                                                 ! Ambient temp leak
         if(tim.lt. 2122.)then
           wleak3 = .21*fact2
           wleak3=wleak3*sqrt((TSAT(prs)+temp+tol)/540.)
           wleak2=wleak3
           wleak1=wleak3
         end if
 C
         GET PPM
 C
```

```
C
        ppm1 = ppm1 + (wleak1 - ppm1 * 1.e - 6 * w) * 1.e 6 / volin 3 / 60.
        ppm2 = ppm2 + (wleak2 - ppm2 * 1.e - 6 * w) * 1.e 6 / volin3 / 60.
        ppm3 = ppm3 + (wleak3 - ppm3 * 1.e - 6 * w) * 1.e 6 / volin 3 / 60.
C
         if (tim.gt.0. .and. tim .ge. tprint) then
           tprint = tim+10.
           write(3) tim,ppm1,ppm2,ppm3,
     1
                         wleak1, wleak2, wleak3,
     2
                         temp, flow, vapor
        end if
C
        goto 200
C
 900
        close(unit=1)
        close(unit=2)
        close(unit=3)
        close(unit=4)
C
        end
C
C
        SUBROUTINE INTRP(A,X,Y)
C
C General purpose interpolation routine. A() is the array to
C interpolate, X in the independent variable, and Y is the
C returned dependent variable. The A() array should be
C configured as follows:
C
C
    a(1)
          Real
                     number of x and y entries (n = 2 * x/y pairs)
C
    a(2)
          Real
                     zero
C
    a(3)
          Real
                     X1
C
    a(4)
          Real
                     Y1
C
    a(5)
          Real
                     X2
C
С
Ca(n+2)
          Real
                     Ym
                        where m=n/2
C Note that both a(1) and a(2) are reset by INTRP after the first
C call, and should not be changed by the calling program
C
        REAL A(*), A1, A2
        INTEGER*4 I1, I2
        EQUIVALENCE (A1, I1), (A2, I2)
C
        A1=A(1)
        A2=A(2)
        IF(I2.NE.0) GOTO 20
                                   ! check if initialized
        I1=A1+1
                                   ! reset to integer
        A(1)=A1
                                   ! and store
        12 = 3
                                   ! set current index at bottom
C
   20
        IF(X-A(I2)) 30,40,50
```

```
IF(I2.EQ.3) GOTO 40    ! at bottom, use first value
  30
                               ! else backup
       I2=I2-2
                               ! and try again
       GOTO 20
С
                               ! use current value
   40
       Y=A(I2+1)
                               ! restore current index
   45
       A(2)=A2
                               ! and done
       RETURN
С
       IF(I2.EQ.I1) GOTO 40   ! at top, use last value
   50
        D1=A(I2+2)-X
        IF(D1.LE.0) GOTO 60 ! not far enough
        Y=A(I2+3) - D1*(A(I2+3)-A(I2+1))/(A(I2+2)-A(I2))
        GOTO 45
С
                               ! move foward
   60
        I2=I2+2
                               ! and try again
        GOTO 20
        END
```

APPROVAL

STS-35 SCRUB 3 HYDROGEN LEAK ANALYSIS

By Dave Seymour

The information in this report has been reviewed for technical content. Review of any information concerning Department of Defense or nuclear energy activities or programs has been made by the MSFC Security Classification Officer. This report, in its entirety, has been determined to be unclassified.

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